

Q1,, (A) Streamlines are not curved

$$(B) \tau_1 = \mu \frac{du}{dy} = \mu \frac{v_1}{\delta} = \tau$$

$$\tau_2 = 2\mu \frac{du}{dy} = 2\mu \frac{v_1}{\delta} = \boxed{2\tau}$$

$$(C) \rho = \frac{M}{V}$$

Mass and Volume both constant

$\therefore \rho$ constant

(Pressure changes)

\Rightarrow The same as at T_1

(D) Specific Weight $\gamma = \rho g$ changes

$$(E) F_{\text{eff}1} = \gamma h_c A = \gamma \left(\frac{h}{2}\right) h w = \frac{1}{2} w \gamma h^2$$

$$F_{\text{eff}2} = \gamma h_c A = \gamma (h) (2hw) = 2w\gamma h^2$$

$$\Rightarrow F_{\text{eff}2} = 4 F_{\text{eff}1} \quad \boxed{4F}$$

Q2₁₁

$$[h] = \frac{[\text{Energy}]}{[\text{Weight}]} = \frac{M L^2 T^{-2}}{M L T^{-2}} = L$$

$$\text{Regime 1} \quad \left[\frac{D^4}{d^4} \frac{v^2}{2g} \right] = \frac{L^4}{L^4} \frac{L^2 T^{-2}}{L T^{-2}} = L$$

⇒ constant 0.04 has no units

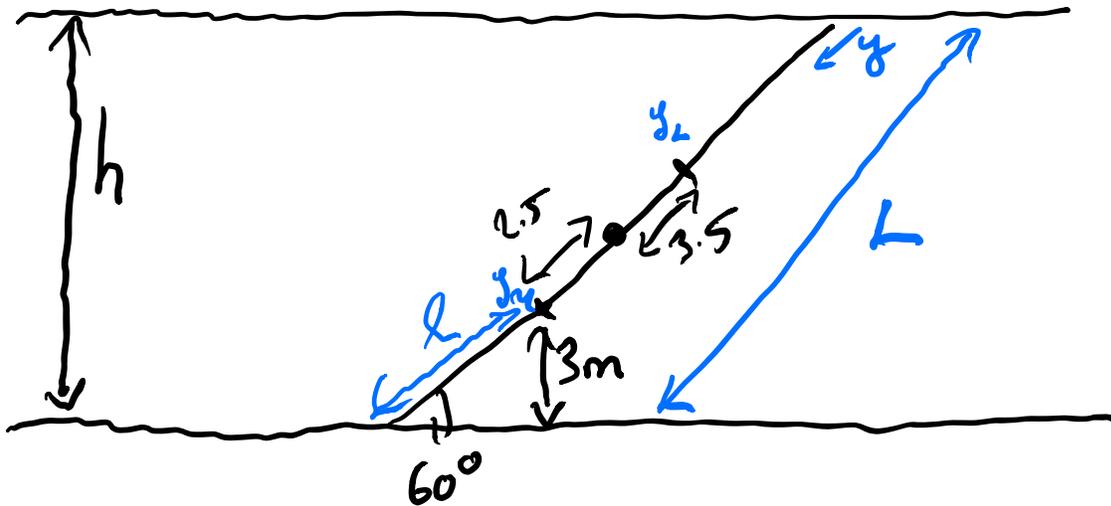
Can be used in any unit system

$$\text{Regime 2} \quad \left[\frac{D^7}{d^7} \frac{v^3}{2g} \right] = \frac{L^7}{L^7} \frac{L^3 T^{-3}}{L T^{-2}} = L^2 T^{-1}$$

⇒ constant 0.7 has units $[L^{-1} T]$

⇒ NOT VALID IN ANY SYSTEM

Q3₁₁



$$h = L \sin 60^\circ$$

$$L = \frac{3}{\sin 60^\circ} \approx 3.5 \text{ m}$$

$$y_{\text{eff}} = \frac{\int_{y_2}^{y_4} y^2 dy}{\int_{y_2}^{y_4} y dy} = \frac{2}{3} \frac{y_4^3 - y_2^3}{y_4^2 - y_2^2}$$

$$y_4 = L - L = L - 3.5$$

$$y_2 = L - L - 6 = L - 9.5$$

$$\therefore y_{\text{eff}} = \frac{2}{3} \frac{(L - 3.5)^3 - (L - 9.5)^3}{(L - 3.5)^2 - (L - 9.5)^2}$$

Also $y_{eff} = L - 2 - 2.5 = L - 6$
 (we want to know when it acts on the hinge)

Setting these two expressions for y_{eff} equal

$$\frac{2}{3} \frac{18L^2 - 234L + 814.5}{12L - 78} = L - 6$$

$$18L^2 - 234L + 814.5 = 18L^2 - 225L + 702$$

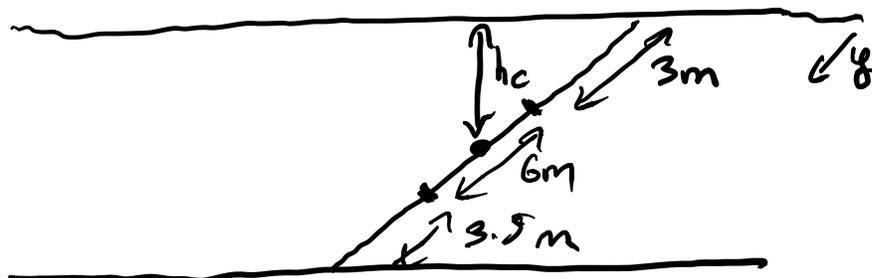
$$112.5 = 9L \Rightarrow \boxed{L = 12.5 \text{ m}}$$

$$h = L \sin 60^\circ \Rightarrow h = 10.8 \text{ m}$$

$$F_{eff} = \gamma h_c A$$

The centroid is at the midpoint of the plate

$$\Rightarrow y_c = 3 + 3 = 6 \text{ m}$$



$$y_c = 6\text{ m} \quad \Rightarrow \quad h_c = y_c \sin 60^\circ \\ = 5.2\text{ m}$$

$$\therefore F_{\text{eff}} = \gamma h_c A \\ = (10000)(5.2)(4 \times 6) \\ = 1248 \text{ kN}$$

Height h does not change with density

\Rightarrow γ_{eff} does not depend on density

However the size of the force will be less.

$$Q4_{//} \quad P_1 + \frac{1}{2} \rho v_1^2 + \rho g z_1 = P_2 + \frac{1}{2} \rho v_2^2 + \rho g z_2$$

$$P_1 = 5 \rho g \quad (\text{manometer equation})$$

$$P_2 = 0 \quad (\text{free jet})$$

$$v_2 = 6 \text{ m/s}$$

$$v_1 = ?$$

$$z_1 = 0$$

$$z_2 = 4$$

$$\therefore 5 \rho g + \frac{1}{2} \rho v_1^2 + 0 = 0 + \frac{1}{2} \rho 36 + 4 \rho g$$

$$\therefore v_1^2 = 36 - 2g = 36 - 20 = 16$$

$$\boxed{v_1 = 4 \text{ m/s}}$$

$$\text{Continuity, } v_1 A_1 = v_2 A_2 \Rightarrow v_1 \frac{\pi}{4} D_1^2 = v_2 \frac{\pi}{4} D_2^2$$

$$\Rightarrow D_2 = \left(\frac{v_1}{v_2} \right)^{1/2} D_1$$

$$= \left(\frac{2}{3} \right)^{1/2} (0.05) = 4.08 \times 10^{-2} \text{ m}$$

$$= \boxed{4.1 \text{ cm}}$$