

Relevant Formulas and Data

$$A = \frac{1}{2}\theta r^2 \quad \text{Area of parabolic section} = \frac{4}{3} \times \text{Area of inscribed triangle.}$$

$$\frac{a_M^3}{T_M^2} = \frac{a_J^3}{T_J^2}$$

$$y = Ax^2 + Bx + C \quad \frac{x^2}{a^2} + \frac{y^2}{b^2} = 1 \quad a = \frac{k}{2} \quad \text{and} \quad b = \sqrt{a^2 - e^2}.$$

$$r = a(1 - \varepsilon \cos \beta) \quad \tan \frac{\alpha}{2} = \sqrt{\frac{1+\varepsilon}{1-\varepsilon}} \tan \frac{\beta}{2} \quad \beta - \varepsilon \sin \beta = \frac{2\pi t}{T}$$

$$\frac{2\pi t}{T} + \varepsilon ()$$

Orbital Data of Planets					
Planet	Semi-major axis in AUs	Period of the orbit in years	Astronomical Eccentricity	Angle of orbital plane with that of the Earth	Average speed in miles/sec
Mercury	0.3871	0.2408	0.2056	7.00°	29.6
Venus	0.7233	0.6152	0.0068	3.39°	21.7
Earth	1.0000	1.0000	0.0167	0.00°	18.5
Mars	1.5237	1.8809	0.0934	1.85°	15.0
Jupiter	5.2028	11.8622	0.0484	1.31°	8.1
Saturn	9.5388	29.4577	0.0557	2.49°	6.0

$$\int_a^b f(x) dx = F(b) - F(a) \quad A = \int_a^b \pi f(x)^2 dx \quad L = \int_a^b \sqrt{1 + f'(x)^2} dx$$

$$\int_a^b cx^r dx = \frac{c}{r+1} x^{r+1} \Big|_a^b = \frac{c}{r+1} b^{r+1} - \frac{c}{r+1} a^{r+1}$$

$$x'(t) = v_0 \cos \varphi_0 \quad \text{and} \quad x(t) = (v_0 \cos \varphi_0) t,$$

$$y'(t) = -gt + v_0 \sin \varphi_0 \quad \text{and} \quad y(t) = -\frac{g}{2}t^2 + (v_0 \sin \varphi_0) t + y_0$$

$$y = \left(\frac{-g}{2v_0^2 \cos^2 \varphi_0} \right) x^2 + (\tan \varphi_0) x + y_0$$

$$\frac{1}{2g} v_0^2 \sin^2 \varphi_0 + y_0 \quad (\text{maximal height attained by the projectile})$$

$$t_{\text{imp}} = \frac{v_0 \sin \varphi_0 + \sqrt{v_0^2 \sin^2 \varphi_0 + 2gy_0}}{g} \quad (\text{time of impact})$$

$$R = \frac{v_0^2}{2g} \sin(2\varphi_0) + \frac{v_0}{g} \sqrt{\frac{v_0^2}{4} \sin^2(2\varphi_0) + 2gy_0 \cos^2 \varphi_0} \quad (\text{range})$$

$$R_{\text{max}} = \frac{v_0^2}{2g} + \frac{v_0}{g} \sqrt{\frac{v_0^2}{4} + gy_0} \quad (\text{maximal range})$$

$$v(t) = \sqrt{v_0^2 + g^2 t^2 - 2g(v_0 \sin \varphi_0)t}$$