

# Dynamics of Unsteady Inviscid and Viscous Detonations in Hydrogen-Air

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50<sup>th</sup> AIAA Aerospace Science Meeting

Nashville, Tennessee

January 11, 2012



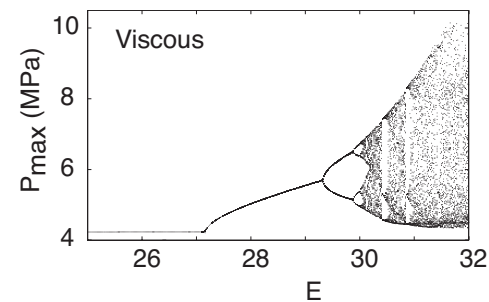
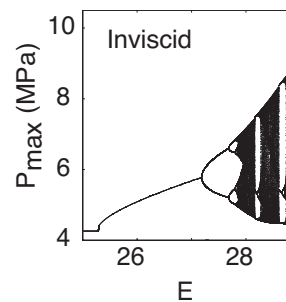
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## Motivation

- Standard result from non-linear dynamics: small scale phenomena can influence large scale phenomena and vice versa.
  - What are the risks of using reactive Euler instead of reactive Navier-Stokes?
  - Might there be risks in using numerical viscosity, LES, and turbulence modeling, all of which filter small scale physical dynamics?
    - For one-step kinetics, yes: there are clear and quantifiable risks.
    - For detailed kinetics, definitive calculations await, but probably yes.
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# Motivation

- It is often argued that viscous forces and diffusive effects are small, do not affect detonation dynamics, and thus can be neglected.
- Tsuboi *et al.*, (*Comb. & Flame*, 2005) report, even when using micron grid sizes, that some structures cannot be resolved.
- Powers, (*JPP*, 2006) showed that two-dimensional detonation patterns are grid-dependent for the reactive Euler equations, but relax to a grid-independent structure for comparable Navier-Stokes calculations.
- Using a one-step kinetics model, we (*49th AIAA ASM*, 2011) showed that when the viscous length scale is similar to that of the finest reaction scale, viscous effects play a critical role in determining the long time behavior of the detonation
- This suggests grid-dependent numerical viscosity may be problematic and one may want to consider the introduction of physical diffusion.



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## Review of hydrogen detonation

- Powers & Paolucci (*AIAA J.*, 2005) studied the reaction length scales of a steady, inviscid hydrogen detonation and found the finest length scales on the order of sub-microns to microns and the largest on the order of centimeters with ambient conditions of  $1 \text{ atm}$  and  $298 \text{ K}$ .
  - These small scales are continuum manifestations of molecular collisions.
  - This range of scales must be resolved to capture the dynamics.
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## Review of hydrogen detonation

- Sussman (Ph.D. Thesis, 1995) performed one-dimensional simulations using only 20 points in the induction zone.
- Using a massively parallel computing environment, Oran *et al.* (*Comb. & Flame*, 1998 ) studied the development of detonation cells in a low-pressure hydrogen mixture in two dimensions.
- Eckett (Ph.D. Thesis, 2001) found that 150 points in the induction zone were necessary capture the dynamics of an overdriven, inviscid detonation at an ambient pressure of 1 *atm*.
- Singh *et al.* (*Comb. Theory & Mod.*, 2001) simulated a one-dimensional, unsteady, viscous, detonation in a hydrogen-oxygen-argon mixture using an adaptive mesh.

## Review of hydrogen detonation

- Yungster and Radhakrishnan (*Comb. Theory & Mod.*, 2004) found that a minimum resolution of near one micron was necessary to capture the dynamics in the inviscid limit at ambient pressure of  $0.197 \text{ atm}$ .
- Daimon and Matsuo (*Phys. Fluids*, 2007) found that as the overdrive is lowered, the long time behavior of the detonation became more complex.
- Using an adaptive mesh in a parallel computing environment, Ziegler *et al.* (*J. Comp. Phys.*, 2011) examined a viscous double-Mach reflection detonation and found that even with a resolution near a micron only qualitative convergence was achieved.

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## **Model: Reactive Navier-Stokes Equations**

- unsteady,
  - detailed mass action kinetics with Arrhenius temperature dependency,
  - ideal mixture of calorically imperfect ideal gases,
  - physical viscosity and thermal conductivity,
  - multicomponent mass diffusion with Soret and DuFour effects
-

# Unsteady, Compressible, Reactive Navier-Stokes Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0,$$

$$\frac{\partial}{\partial t} (\rho \mathbf{u}) + \nabla \cdot (\rho \mathbf{u} \mathbf{u} + p \mathbf{I} - \boldsymbol{\tau}) = \mathbf{0},$$

$$\frac{\partial}{\partial t} \left( \rho \left( e + \frac{\mathbf{u} \cdot \mathbf{u}}{2} \right) \right) + \nabla \cdot \left( \rho \mathbf{u} \left( e + \frac{\mathbf{u} \cdot \mathbf{u}}{2} \right) + (p \mathbf{I} - \boldsymbol{\tau}) \cdot \mathbf{u} + \mathbf{q} \right) = 0,$$

$$\frac{\partial}{\partial t} (\rho Y_i) + \nabla \cdot (\rho \mathbf{u} Y_i + \mathbf{j}_i) = \overline{M}_i \dot{\omega}_i,$$

$$p = \mathcal{R} T \sum_{i=1}^N \frac{Y_i}{M_i}, \quad e = e(T, Y_i), \quad \dot{\omega}_i = \dot{\omega}_i(T, Y_i),$$

$$\mathbf{j}_i = \rho \sum_{\substack{k=1 \\ k \neq i}}^N \frac{\overline{M}_i D_{ik} Y_k}{\overline{M}} \left( \frac{\nabla y_k}{y_k} + \left( 1 - \frac{\overline{M}_k}{\overline{M}} \right) \frac{\nabla p}{p} \right) - \frac{D_i^T \nabla T}{T},$$

$$\boldsymbol{\tau} = \mu \left( \nabla \mathbf{u} + (\nabla \mathbf{u})^T - \frac{2}{3} (\nabla \cdot \mathbf{u}) \mathbf{I} \right),$$

$$\mathbf{q} = -k \nabla T + \sum_{i=1}^N \mathbf{j}_i h_i - \mathcal{R} T \sum_{i=1}^N \frac{D_i^T}{M_i} \left( \frac{\nabla \bar{y}_i}{\bar{y}_i} + \left( 1 - \frac{\overline{M}_i}{\overline{M}} \right) \frac{\nabla p}{p} \right).$$



# Computational Methods

- Inviscid Dynamics
  - High-order shock-fitting algorithm adapted from Henrick *et al.* (*J. Comp. Phys.*, 2006).
  - Equations transformed to a shock-attached frame, jump conditions enforced at shock boundary, and fifth order Runge-Kutta used for time integration.
- Viscous Dynamics
  - Wavelet Adaptive Multiresolution Representation (WAMR) method first developed by Vasilyev and Paolucci (*J. Comp. Phys.*, 1996, 1997) employed.
  - An adaptive mesh refinement technique using wavelet functions which have compact support in both space and time enables the use of many less points to accurately represent a flow field.

## Case Examined

- Overdriven detonations with ambient conditions of  $0.421 \text{ atm}$  and  $293.15 \text{ K}$
- Initial stoichiometric mixture of  $2H_2 + O_2 + 3.76N_2$
- $D_{CJ} \sim 1961 \text{ m/s}$
- Overdrive is defined as  $f = D_o^2 / D_{CJ}^2$
- Overdrives of  $1.025 < f < 1.150$  were examined

## Continuum Scales

- The mean-free path scale is the cut-off minimum length scale associated with continuum theories.
- A simple estimate for this scale is given by Vincenti and Kruger (1967):

$$\lambda = \frac{\overline{M}}{\sqrt{2}\pi\mathcal{N}_A\rho d^2} \sim \mathcal{O}(10^{-6} \text{ cm}). \quad (1)$$

- The finest reaction length scale is  $L_r \sim \mathcal{O}(10^{-4} \text{ cm})$ .
- A simple estimate of a viscous length scale is:

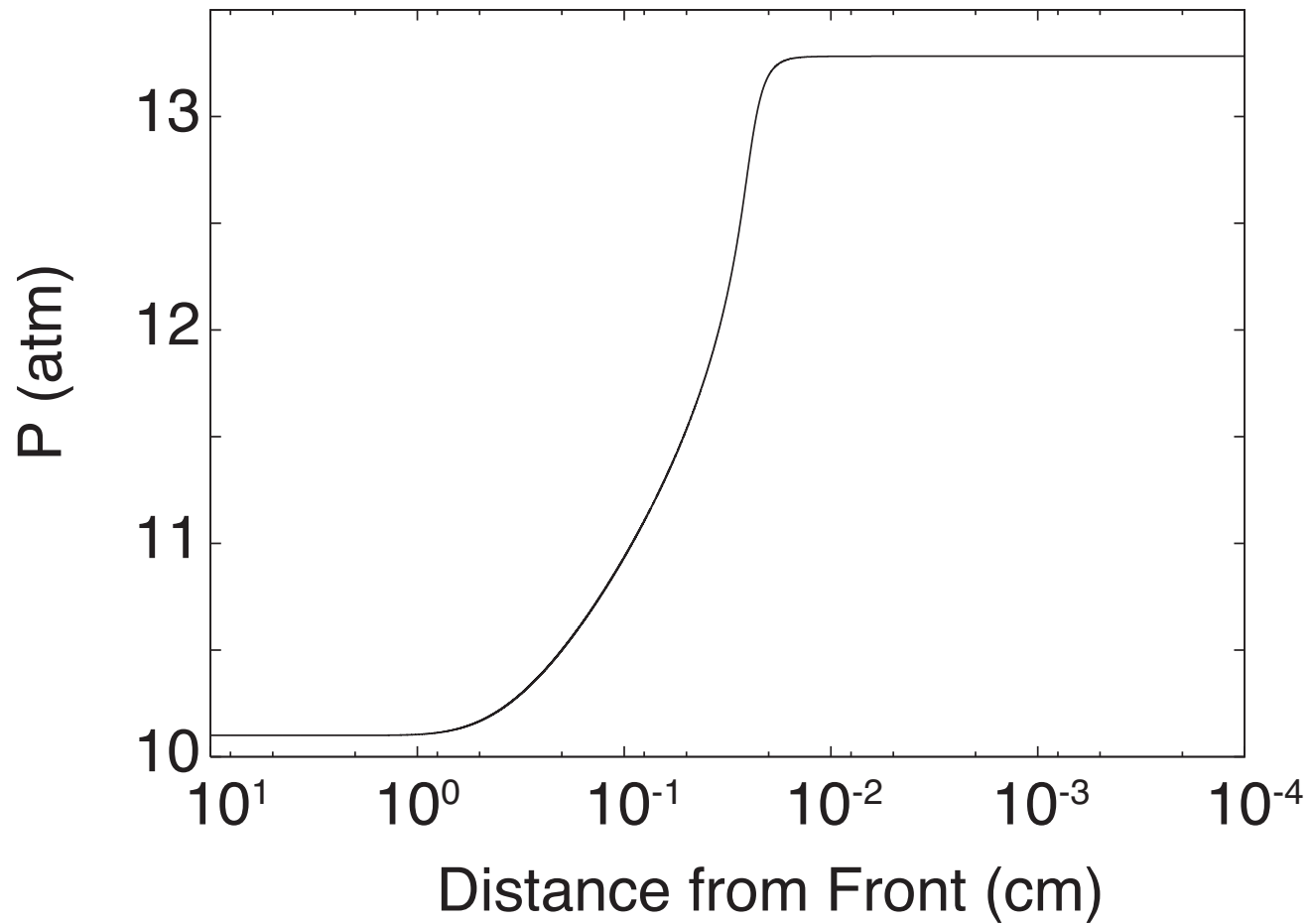
$$L_\mu = \frac{\nu}{c} = \frac{6 \times 10^{-1} \text{ cm}^2/\text{s}}{9 \times 10^4 \text{ cm}/\text{s}} \sim \mathcal{O}(10^{-5} \text{ cm}). \quad (2)$$

- $\lambda < L_\mu < L_r$



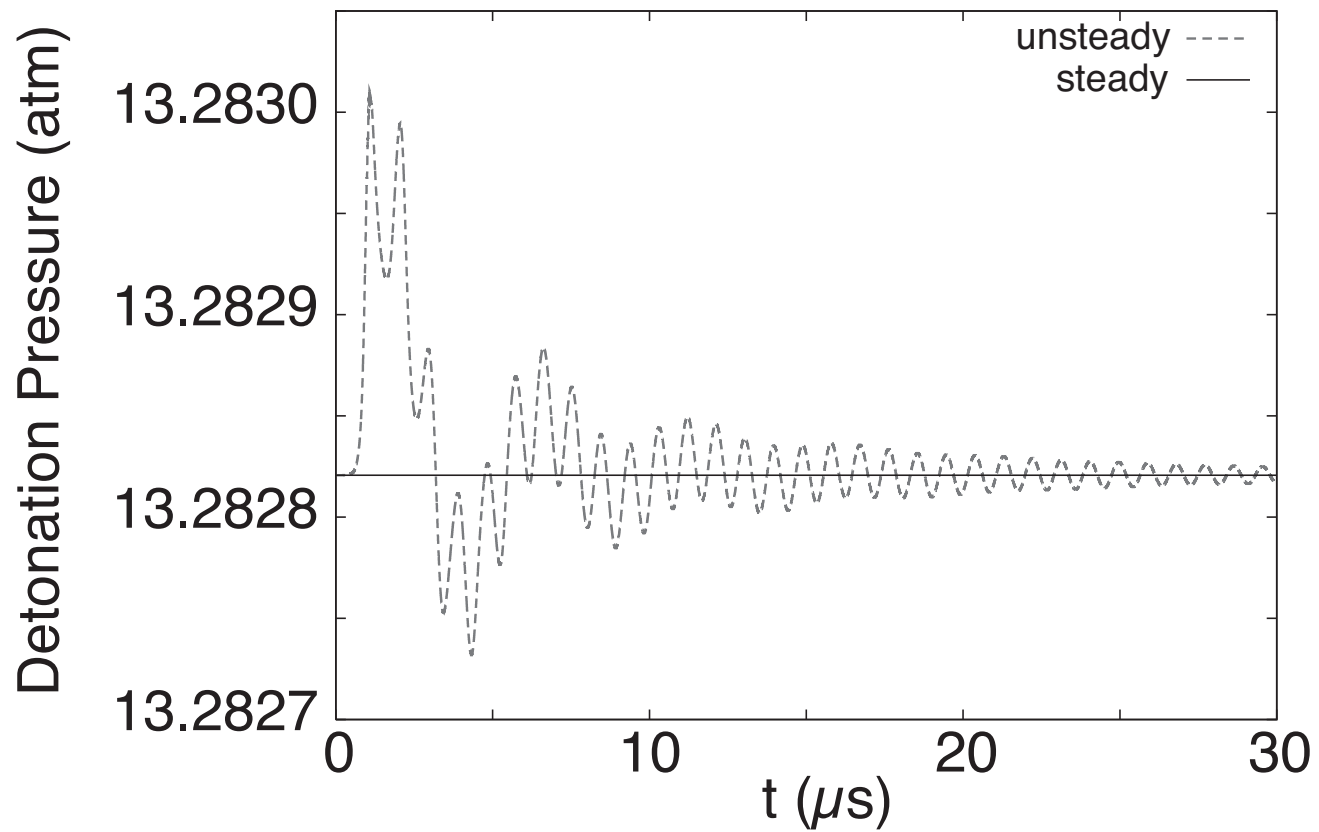
# Inviscid Steady-State: Pressure

$$f = 1.15$$

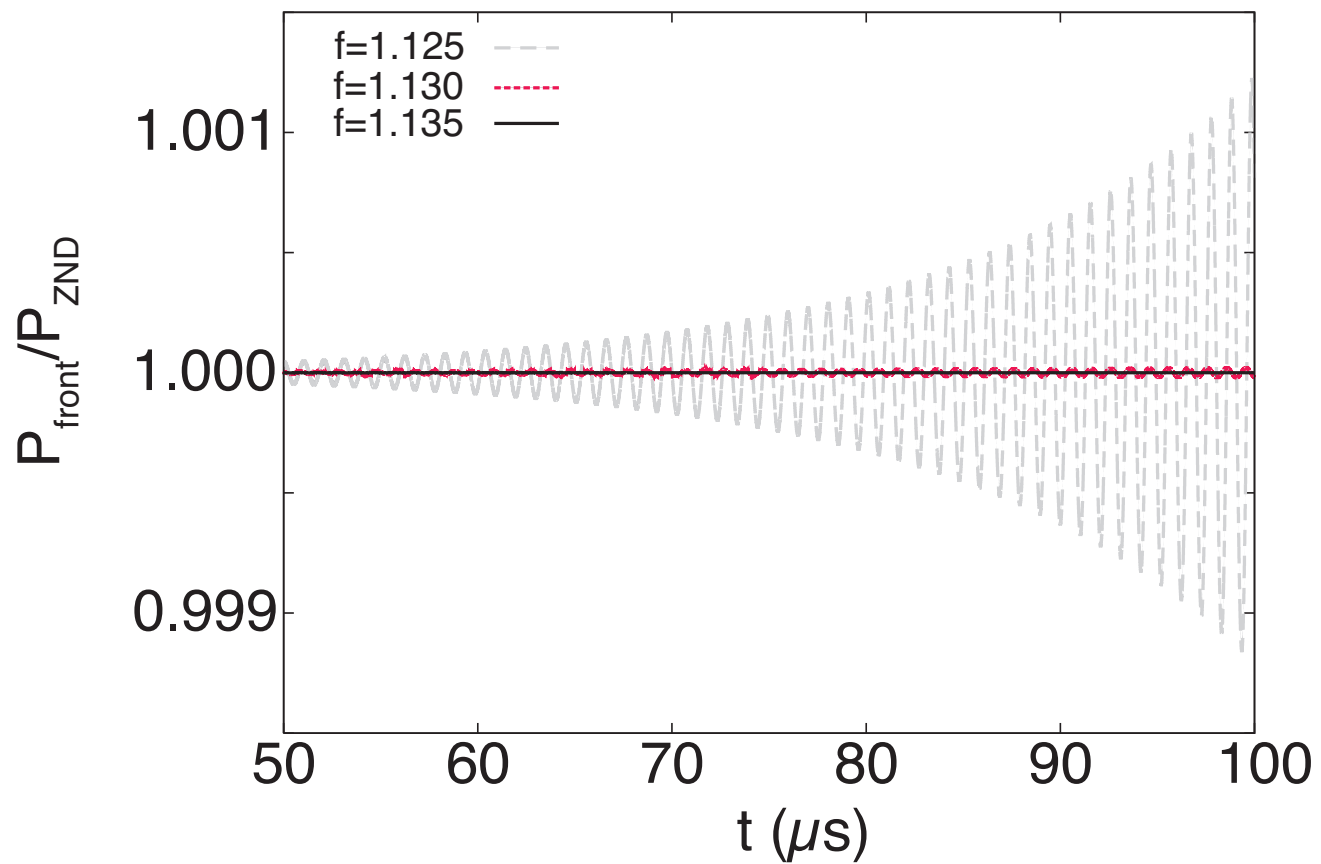


# Inviscid Transient Behavior: Stable Detonation

$$f = 1.15$$

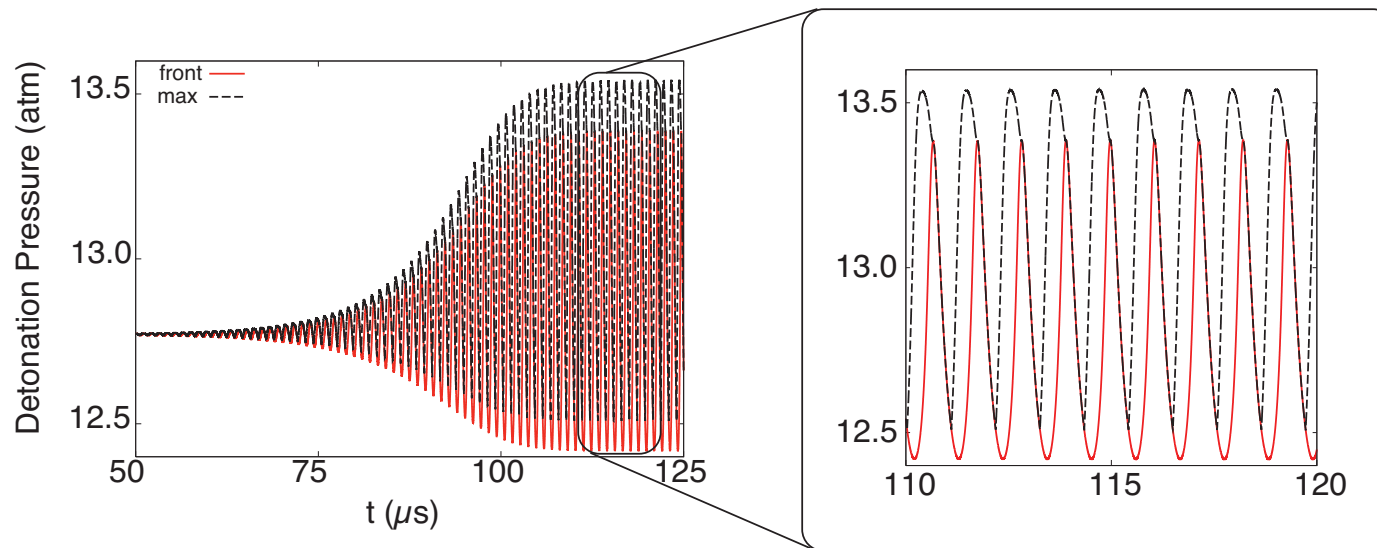


# Near Neutral Stability



# Inviscid Transient Behavior: Unstable Detonation

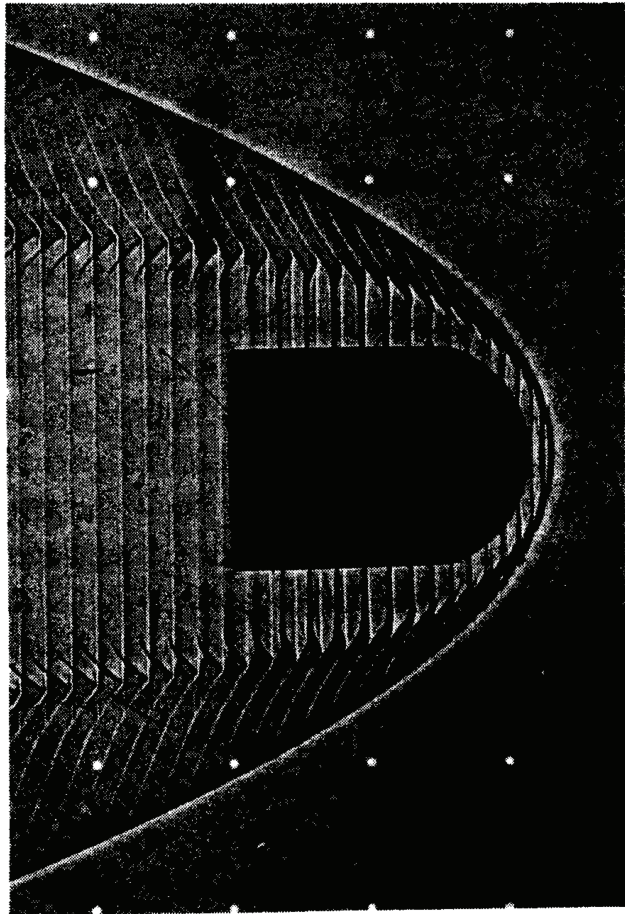
$$f = 1.10$$



- Frequency of  $0.97 \text{ MHz}$  agrees well with both the frequency,  $1.04 \text{ MHz}$ , observed by Lehr (*Astro. Acta*, 1972) in experiments and the frequency,  $1.06 \text{ MHz}$ , predicted by Yungster and Radhakrishnan.
- The maximum detonation front pressure predicted,  $13.5 \text{ atm}$ , is similar to the value of  $14.0 \text{ atm}$  found by Daimon and Matsuo.



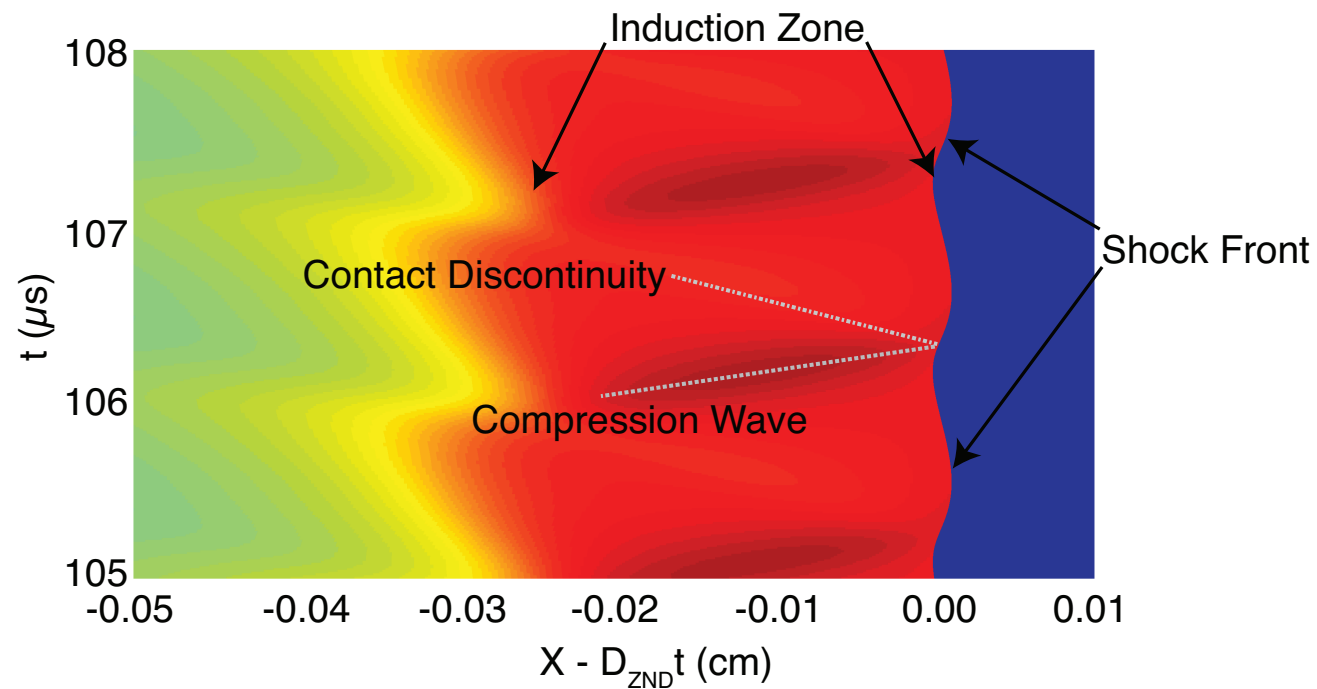
## Lehr's High Frequency Instability



- Experiment of shock-induced combustion in flow around a projectile in an ambient stoichiometric mixture of  $2H_2 + O_2 + 3.76N_2$  at  $0.421 \text{ atm}$ .
- Projectile velocity yields an equivalent overdrive of  $f \approx 1.1$
- The observed frequency was approximately  $1.04 \text{ MHz}$

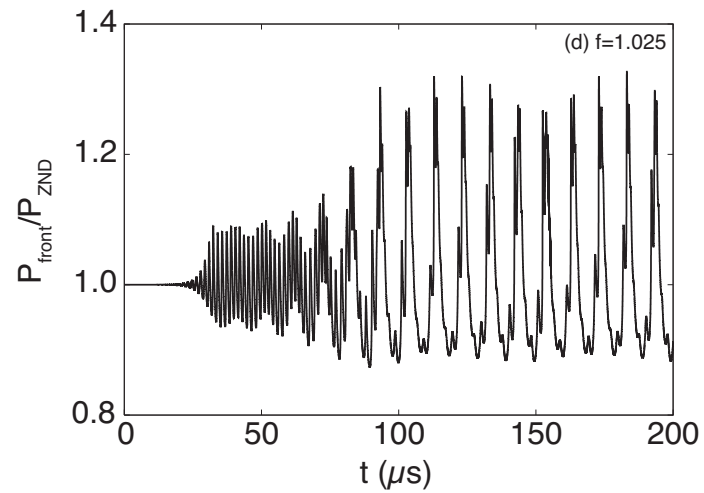
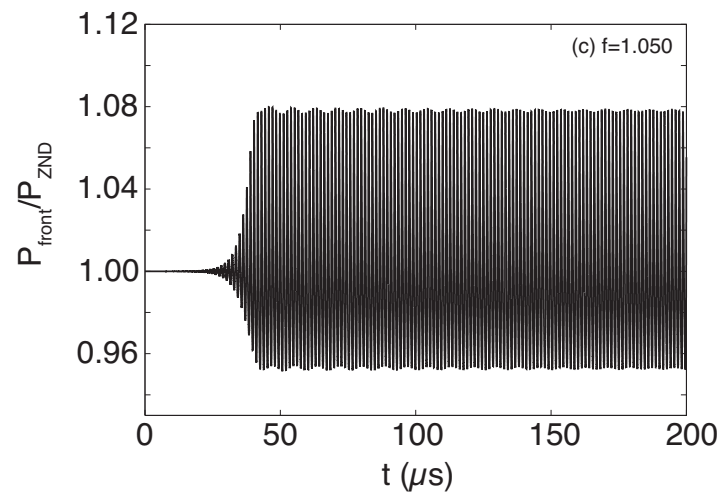
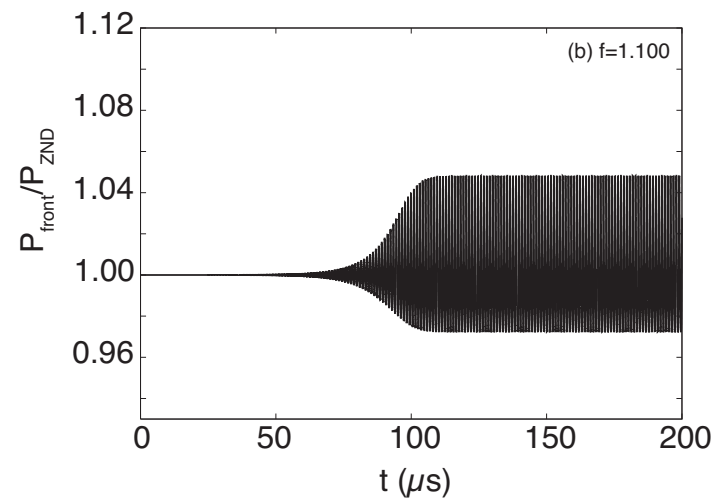
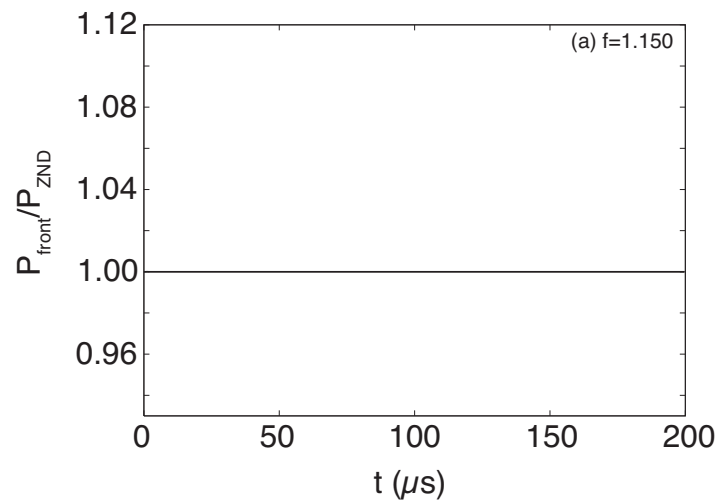
# Unstable, Inviscid Detonation: $x-t$ Diagram

$$f = 1.10$$

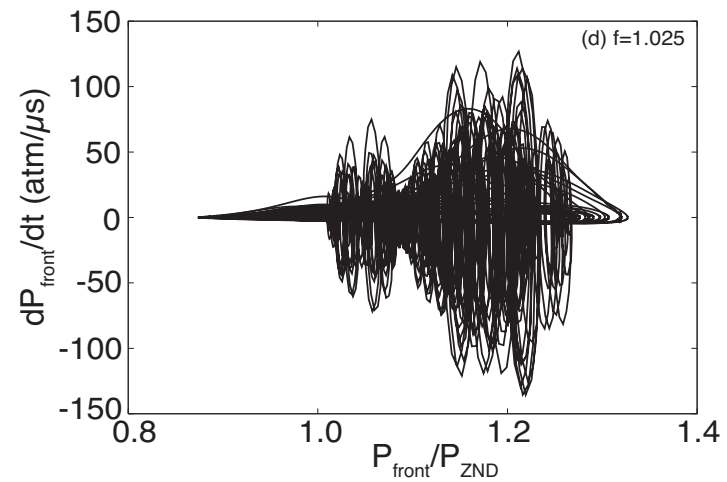
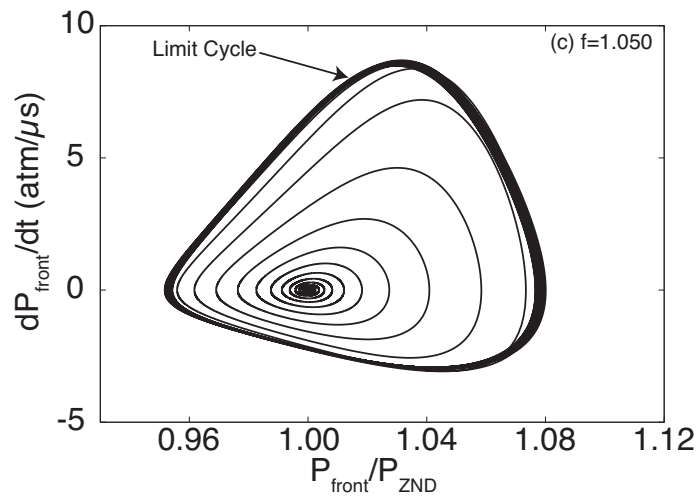
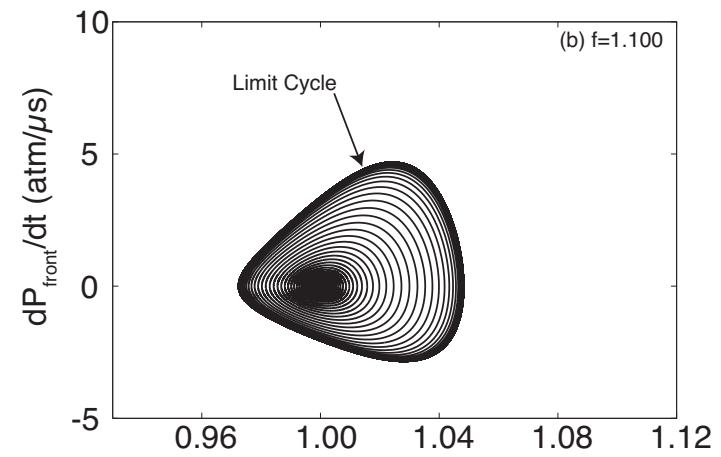
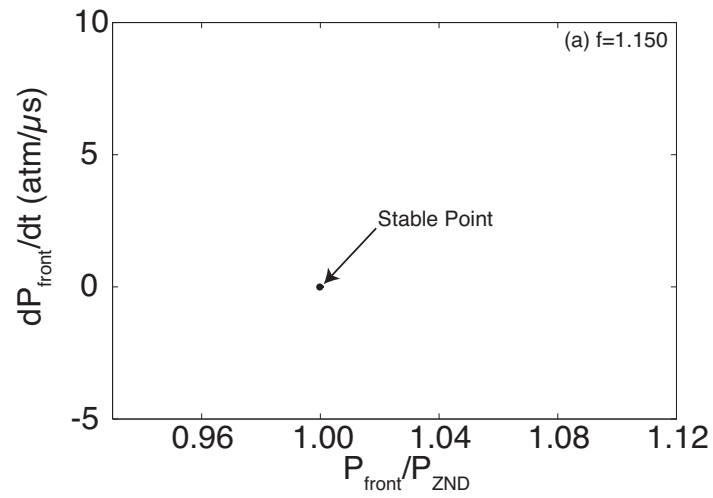


A  $x-t$  diagram of density in a Galilean reference frame traveling at  $2057 \text{ m/s}$ .

# Inviscid Transient Behavior: Various Overdrives

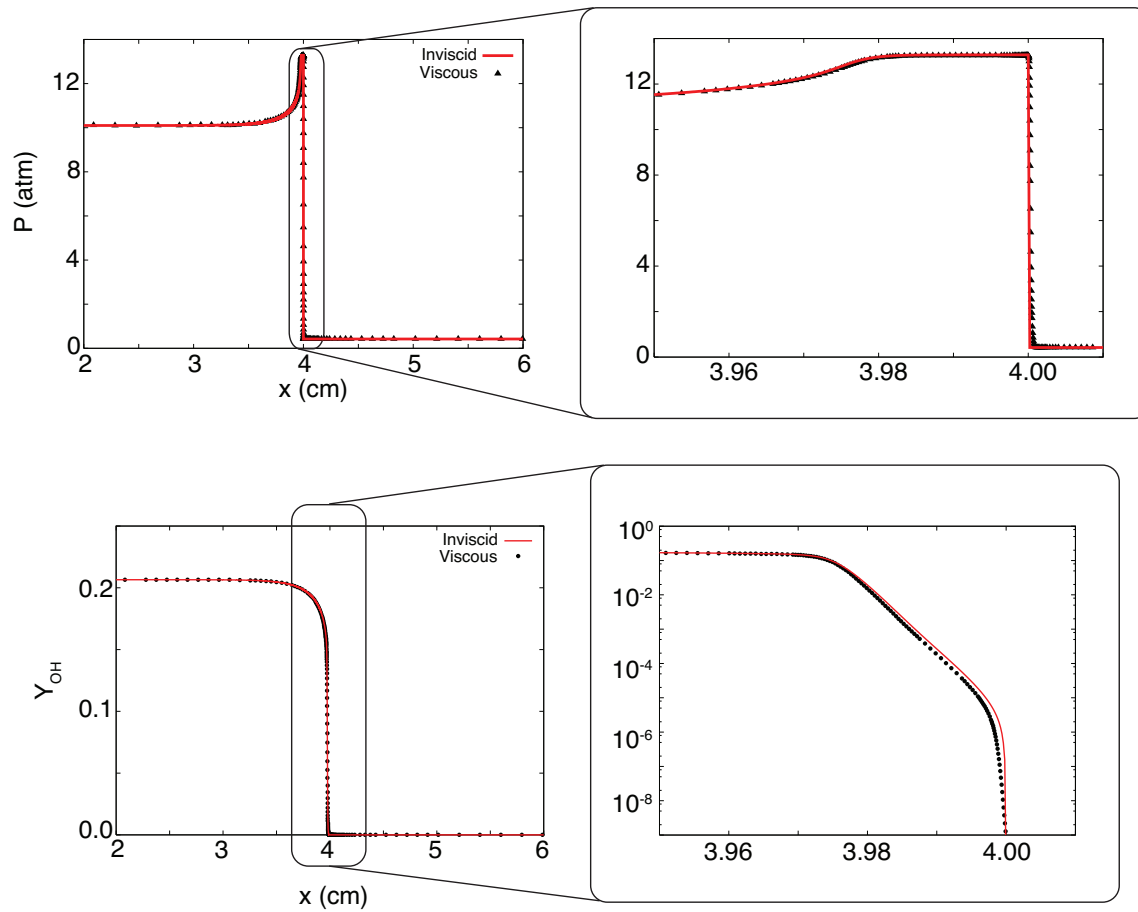


# Inviscid Phase Portraits: Various Overdrives



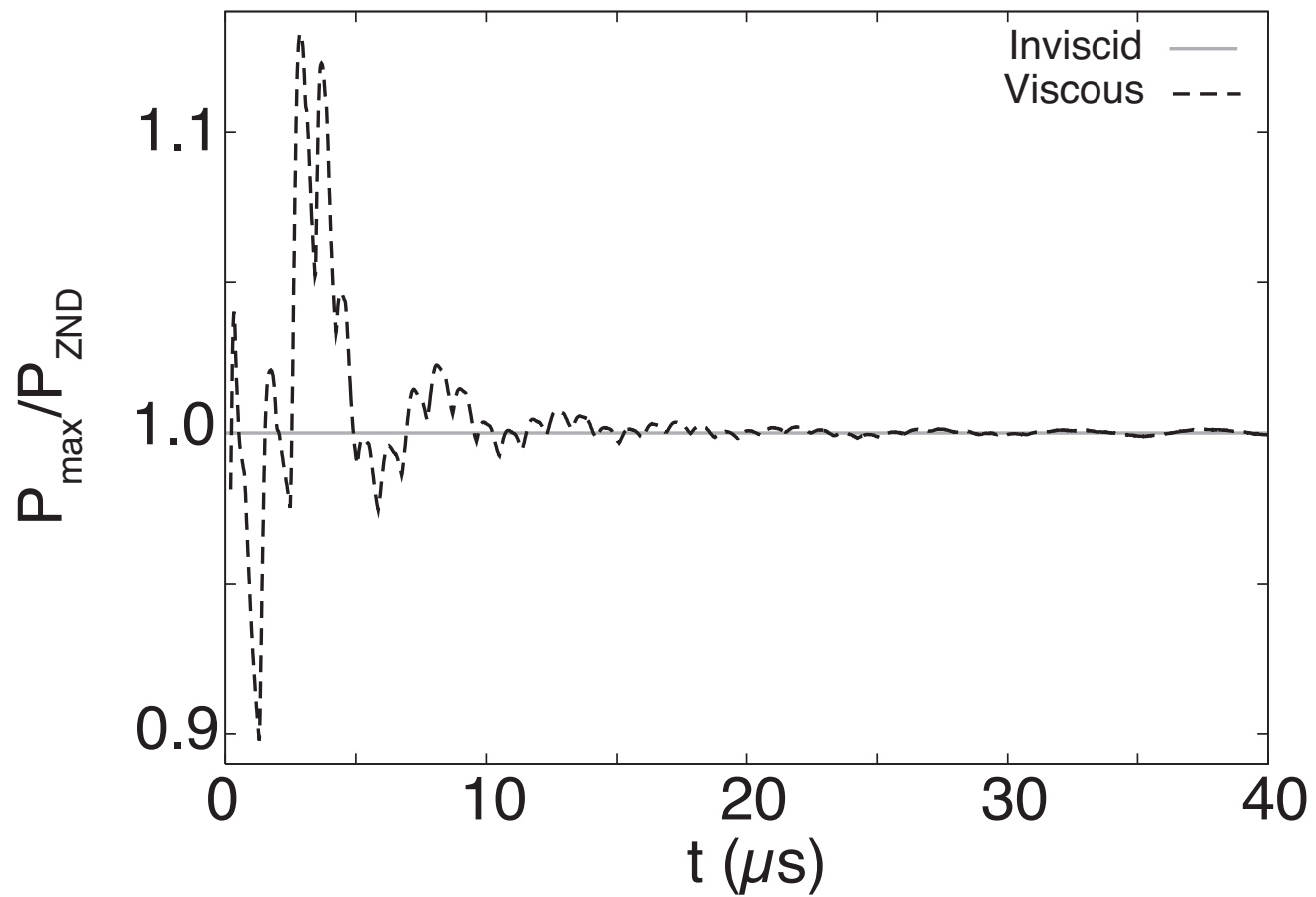
# Stable, Viscous Detonation: Long Time Structure

$$f = 1.15$$



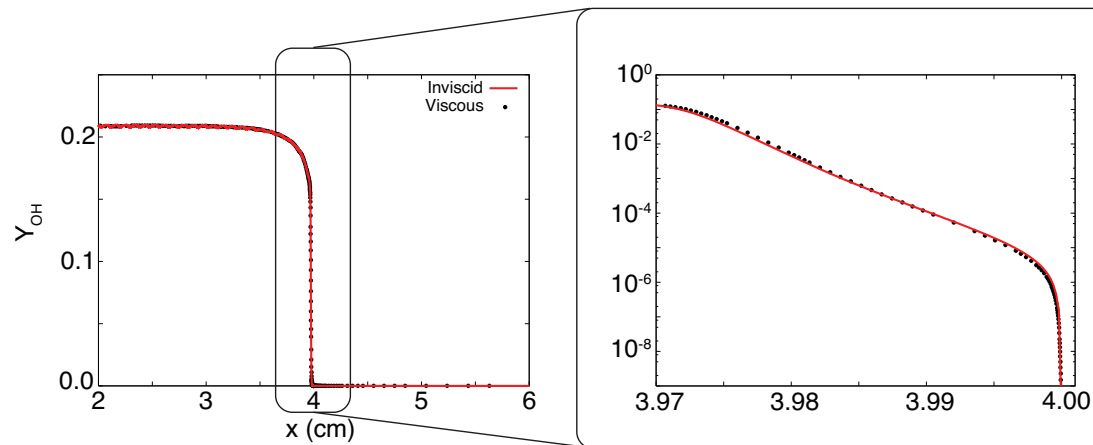
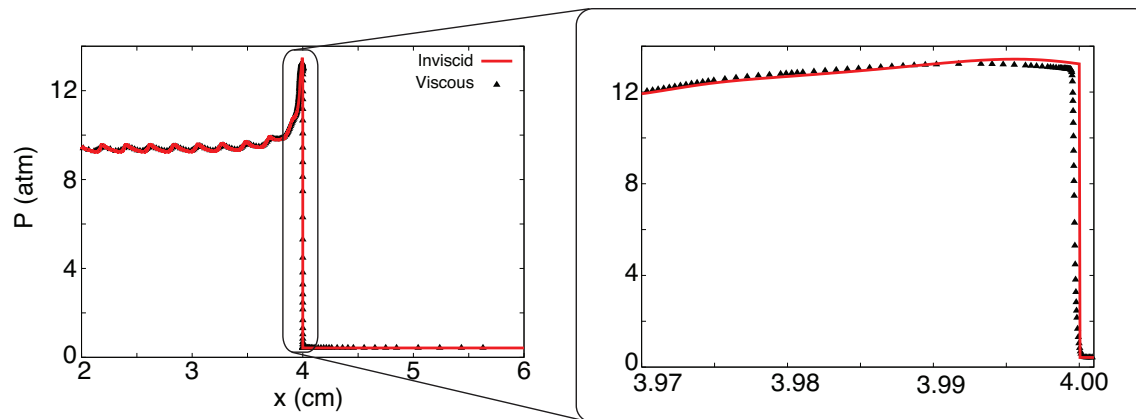
# Stable, Viscous Detonation: Transient Behavior

$$f = 1.15$$



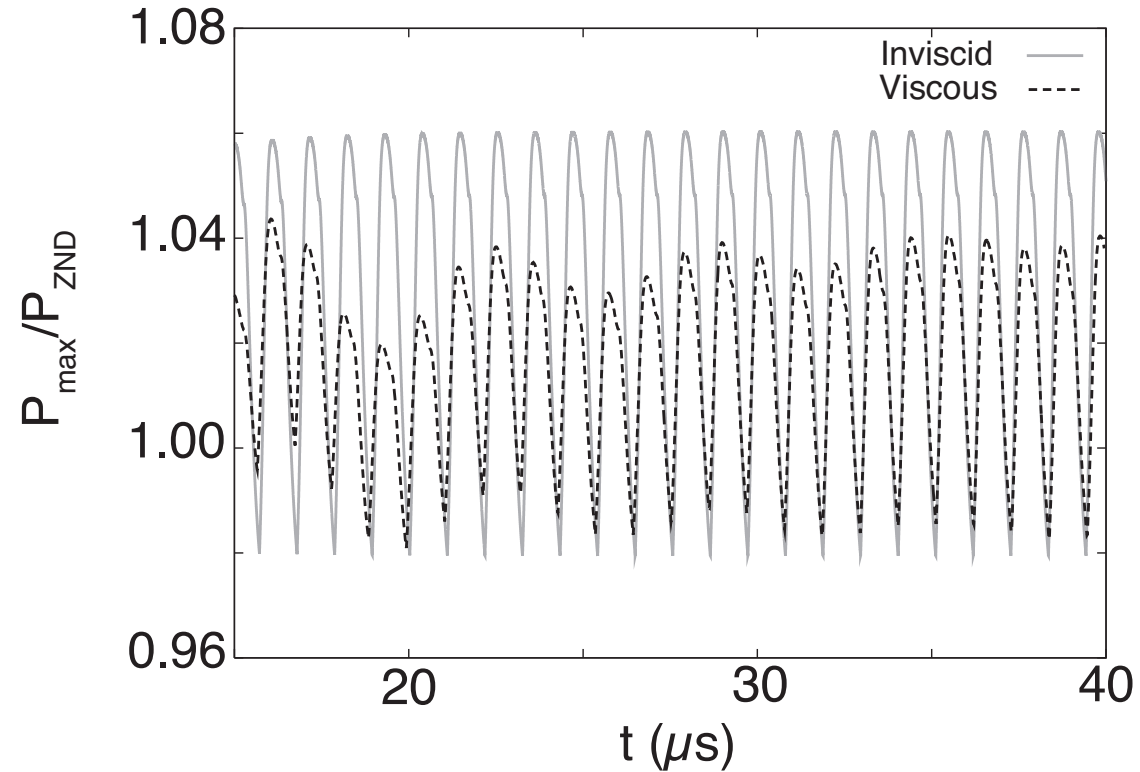
# Unstable, Viscous Detonation: Long Time Structure

$$f = 1.10$$



# Unstable, Viscous Detonation: Transient Behavior

$$f = 1.10$$

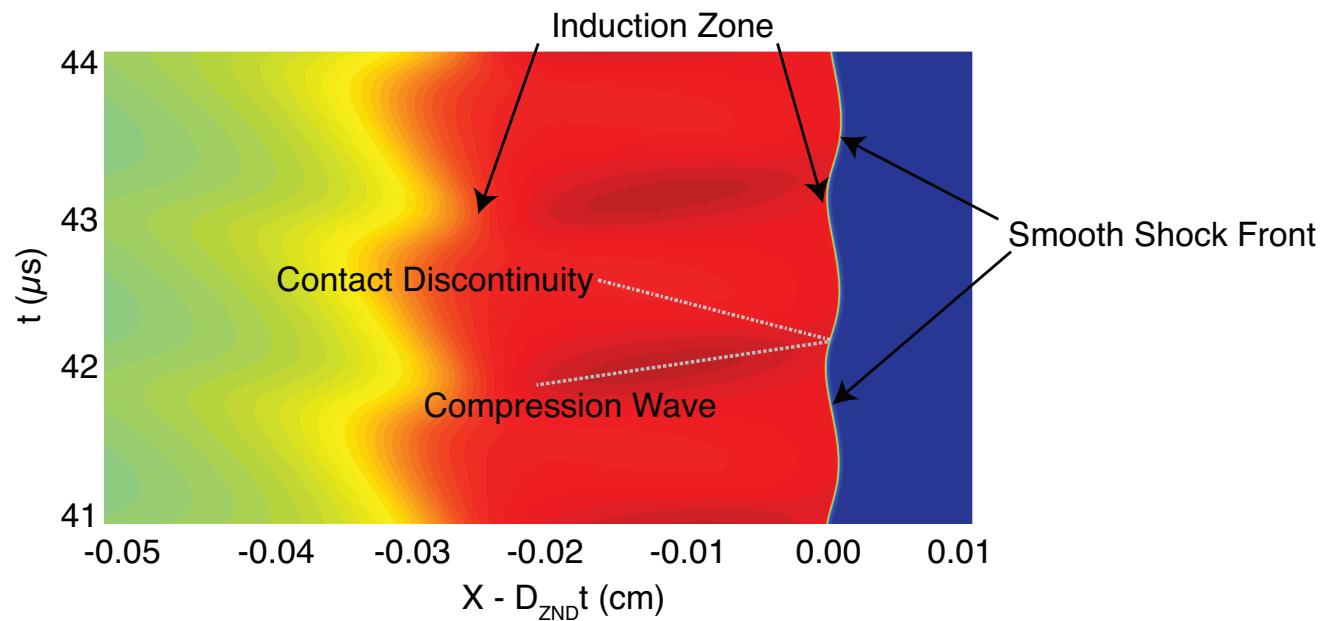


The addition of viscous effects have a stabilizing effect, decreasing the amplitude of the oscillations by  $\sim 25\%$ .



# Unstable, Viscous Detonation: $x-t$ Diagram

$$f = 1.10$$



A  $x-t$  diagram of density in a Galilean reference frame traveling at  $2057 \text{ m/s}$ .

## Conclusions

- Unsteady, inviscid detonation dynamics can be accurately simulated when all reaction length scales admitted by detailed kinetics are fully resolved using a fine grid; the shock-fitting technique used assures numerical viscosity is minimal.
- At high overdrives, the detonations are stable.
- As the overdrive is decreased, the long time behavior becomes progressively more complex.
- In the inviscid limit a critical overdrive,  $f = 1.130$ , is found below which oscillations at a single frequency appear.
- As the overdrive is lowered, the amplitude of these oscillations increases.

## Conclusions

- Lowering the overdrive yet further gives rise to oscillations at multiple frequencies.
- The predicted  $0.97 \text{ MHz}$  frequency for a  $f = 1.10$  overdriven detonation agrees well with the frequency of  $1.04 \text{ MHz}$  observed by Lehr in his experiments of shock-induced combustion flow around spherical projectiles.
- The structure of the overdriven detonation relative to the inviscid limit is modulated by the addition of mass, momentum, and energy diffusion.
- The addition of viscous effects has a stabilizing effect on the long time behavior of a detonation; the amplitude of the oscillations is significantly reduced.