

# Harmony in High Speed Combustion

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ACMS Applied Math Seminar

5 February 2015



## Acknowledgments

- Christopher M. Romick, Ph.D. candidate, ND-AME
- Tariq D. Aslam, Technical Staff Member, LANL
- Romick, Aslam, Powers, 2012, The effect of diffusion on the dynamics of unsteady detonation, *Journal of Fluid Mechanics*, 699:453-464.
- Romick, Aslam, Powers, 2013, On the resolution necessary to capture dynamics of unsteady detonation, 51st AIAA Aerospace Sciences Meeting and Exhibit, AIAA 2013-0887.
- Romick, Aslam, Powers, 2015, Verified and validated calculation of unsteady dynamics of viscous hydrogen-air detonation, *Journal of Fluid Mechanics*, to appear.

## Verification and Validation Overview

- We will consider here *verification* and *validation* of a tough multi-scale problem using *Direct Numerical Modeling* which captures both coarse and fine scales.
- One key algorithm is the *Wavelet Adaptive Multiresolution Method* (WAMR), one of the main codes in the University of Notre Dame-led *Center for Shock Wave Processing of Advanced Materials* (C-SWARM), a NNSA-supported PSAAP II Center.
- C-SWARM is in Year 1 of a five-year project to prepare for scientific computing in an exascale environment of challenging multi-scale shock physics problems.

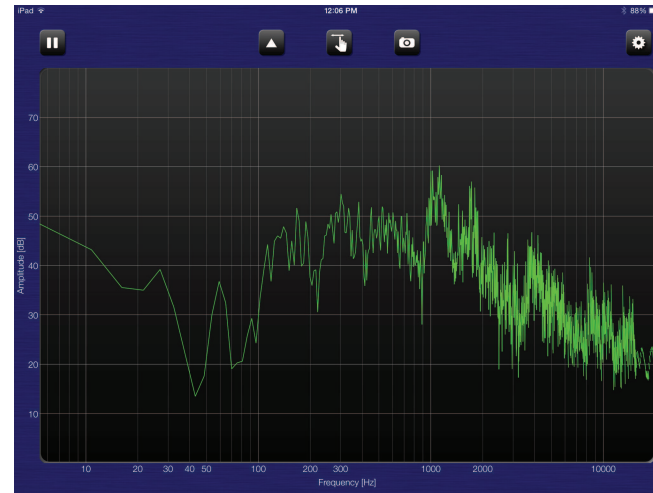
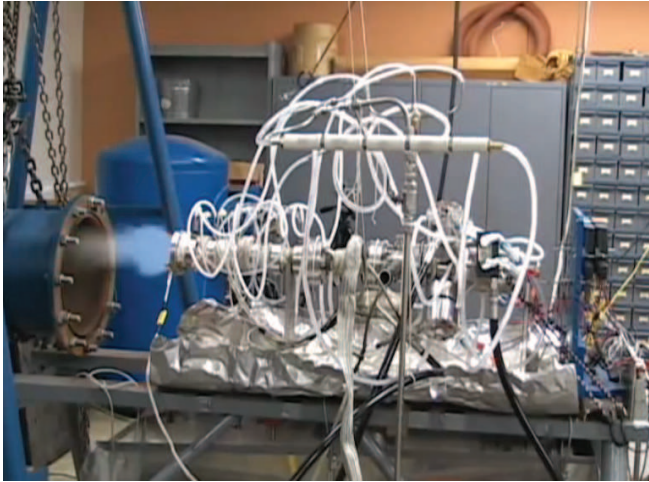
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## **Verification and Validation Overview, cont.**

- Joint effort with Notre Dame, Indiana U., and Purdue U.
  - Our problem is shocking mechanically pre-activated pressed metallic powders to synthesize new metallic structures.
  - We will develop verified and validated predictive codes prepared for an exascale environment.
  - The WAMR code, in development at Notre Dame for 20 years, will be used today on a different problem.
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# Disharmony in High Speed Combustion



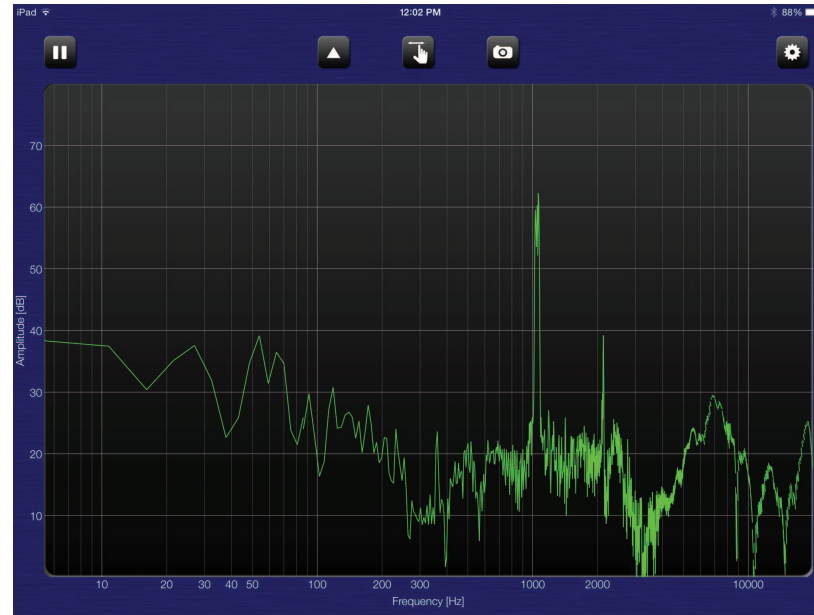
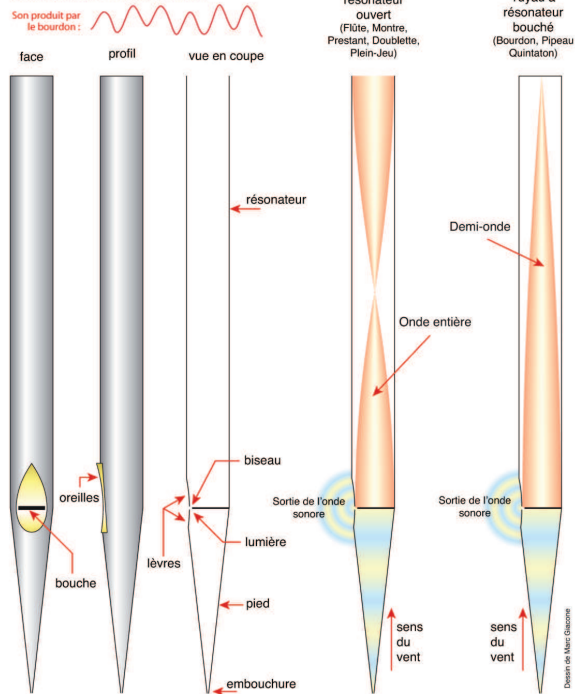
<https://www.youtube.com/watch?v=rYxsilgRxi4>  
Prof. Frank Lu, University of Texas-Arlington.

Described as “25 Hz,” but there is acoustic energy present across the frequency spectrum. Disorder.

# Harmony: Organ Pipe Resonance

## Comment se produit le son dans un tuyau d'orgue. LE TUYAU À BOUCHE

Un tuyau à résonateur ouvert produit une onde entière.  
Un tuyau à résonateur bouché produit une demi-onde,  
si bien qu'il parle une octave plus bas qu'un tuyau de  
même hauteur à résonateur ouvert.



$a/l \sim 1000$  Hz. Higher order harmonic at 2000 Hz. Order.

## Harmony in Low Speed Combustion?



<http://www.youtube.com/watch?v=w5zWkSuYf1Y>  
Order.

## Motivation

- Combustion dynamics are influenced by various balances of *advection, reaction, and diffusion*.
- Depending on physical flow conditions, one may observe and predict simple structures, patterned harmonic structures, or chaotic structures.
- Often, the critical balance is between *advection* and *reaction* with diffusion serving as only a small perturbation.
- Near stability thresholds, diffusion can play a determining role.
- Full non-linear dynamics can induce complex behavior.
- Extreme care *may or may not be* needed in numerical simulation to carefully capture the multi-scale physics.

## Introduction

- Standard result from non-linear dynamics: small scale phenomena can influence large scale phenomena and vice versa.
- What are the risks of using models which ignore diffusion (Euler vs. Navier-Stokes)?
- Might there be risks in using implicit time-advancement, numerical viscosity, LES, and turbulence modeling, all of which introduce *nonphysical diffusion* to filter small scale physical dynamics?

## Introduction-Continued

- Powers & Paolucci (*AIAA J*, 2005) studied the reaction length scales of inviscid  $\text{H}_2\text{-O}_2$  detonations and found the finest length scales on the order of sub-microns to microns and the largest on the order of centimeters for atmospheric ambient pressure.
- This range of scales must be resolved to capture the dynamics.
- In a one-step kinetic model only a single length scale is induced compared to the multiple length scales of detailed kinetics.
- We examine i) a simple one-step model and ii) a detailed model appropriate for hydrogen.

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# One-Step Reaction Kinetics Model

## One-Dimensional Unsteady Compressible Reactive Navier-Stokes Equations

$$\frac{\partial \rho}{\partial t} + \frac{\partial}{\partial x} (\rho u) = 0,$$

$$\frac{\partial}{\partial t} (\rho u) + \frac{\partial}{\partial x} (\rho u^2 + P - \tau) = 0,$$

$$\frac{\partial}{\partial t} \left( \rho \left( e + \frac{u^2}{2} \right) \right) + \frac{\partial}{\partial x} \left( \rho u \left( e + \frac{u^2}{2} \right) + j^q + (P - \tau) u \right) = 0,$$

$$\frac{\partial}{\partial t} (\rho Y_B) + \frac{\partial}{\partial x} (\rho u Y_B + j_B^m) = \rho r.$$

Equations are transformed to a steady moving reference frame.



## Constitutive Relations

$$P = \rho RT,$$

$$e = \frac{p}{\rho(\gamma - 1)} - qY_B,$$

$$r = H(P - P_s)a(1 - Y_B) e^{-\frac{\tilde{E}}{p/\rho}},$$

$$j_B^m = -\rho \mathcal{D} \frac{\partial Y_B}{\partial x},$$

$$\tau = \frac{4}{3} \mu \frac{\partial u}{\partial x},$$

$$j^q = -k \frac{\partial T}{\partial x} + \rho \mathcal{D} q \frac{\partial Y_B}{\partial x}.$$

with  $D = 10^{-4} \frac{\text{m}^2}{\text{s}}$ ,  $k = 10^{-1} \frac{\text{W}}{\text{mK}}$ , and  $\mu = 10^{-4} \frac{\text{Ns}}{\text{m}^2}$ , so for  $\rho_o = 1 \frac{\text{kg}}{\text{m}^3}$ ,  
 $Le = Sc = Pr = 1$ .

## Case Examined

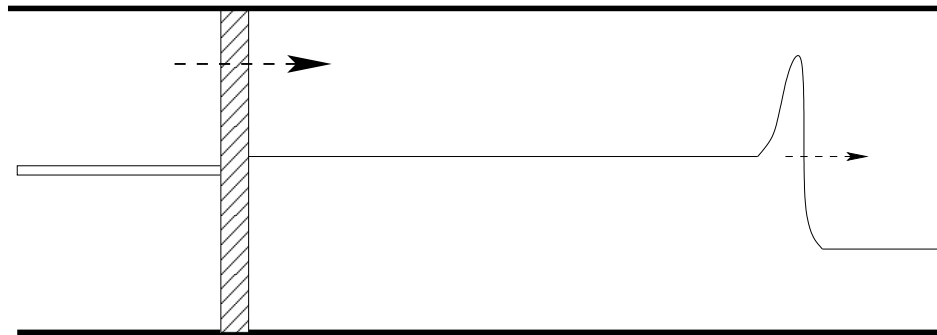
Let us examine this one-step kinetic model with:

- a fixed reaction length,  $L_{1/2} = 10^{-6}$  m, which is similar to that of H<sub>2</sub>-O<sub>2</sub>.
- a fixed the diffusion length,  $L_{\mu} = 10^{-7}$  m; mass, momentum, and energy diffusing at the same rate.
- an ambient pressure,  $P_o = 101325$  Pa, ambient density,  $\rho_o = 1$  kg/m<sup>3</sup>, heat release  $q = 5066250$  m<sup>2</sup>/s<sup>2</sup>, and  $\gamma = 6/5$ .

## Numerical Method

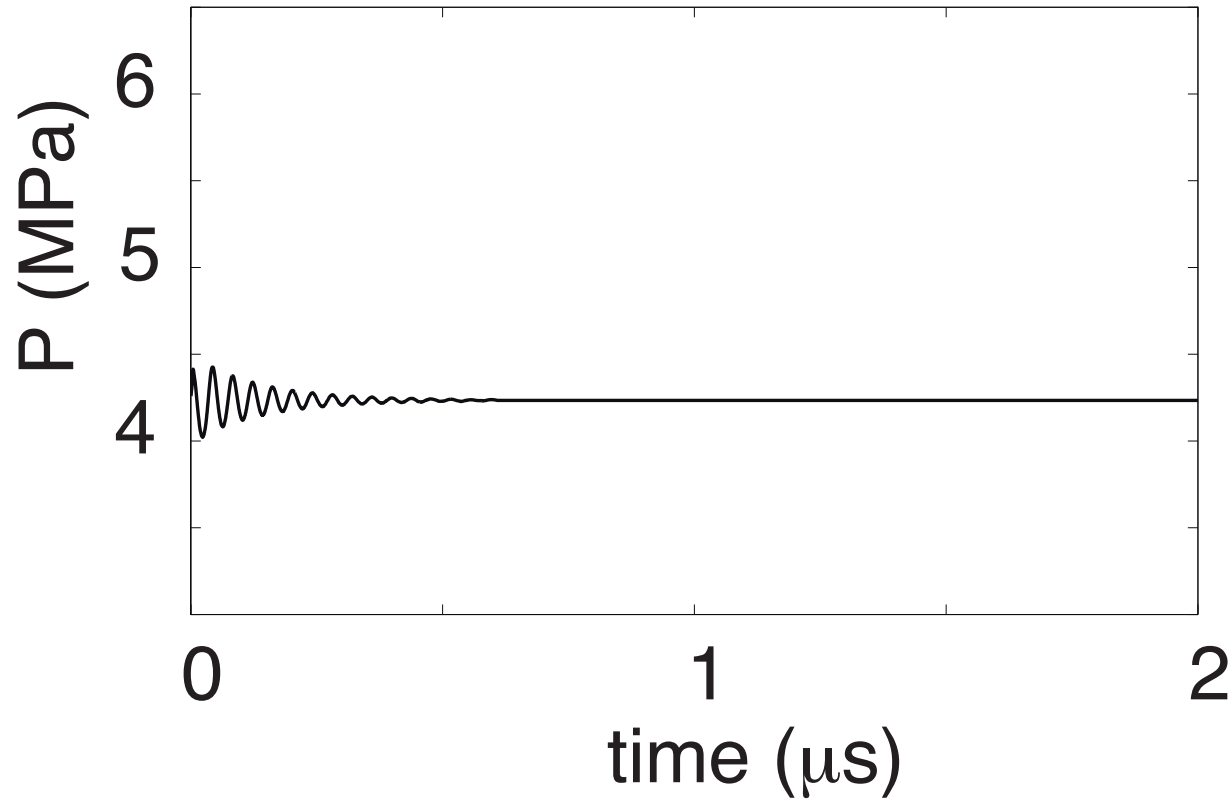
- Finite difference, uniform grid  
( $\Delta x = 2.50 \times 10^{-8}$  m,  $N = 8001$ ,  $L = 0.2$  mm) .
- Computation time = 192 hours for 10  $\mu$ s on an AMD 2.4 GHz with 512 kB cache.
- A point-wise method of lines approach was used.
- Advective terms were calculated using a combination of fifth order WENO and Lax-Friedrichs.
- Sixth order central differences were used for the diffusive terms.
- Temporal integration was accomplished using a third order Runge-Kutta scheme.

## Physical Piston Problem

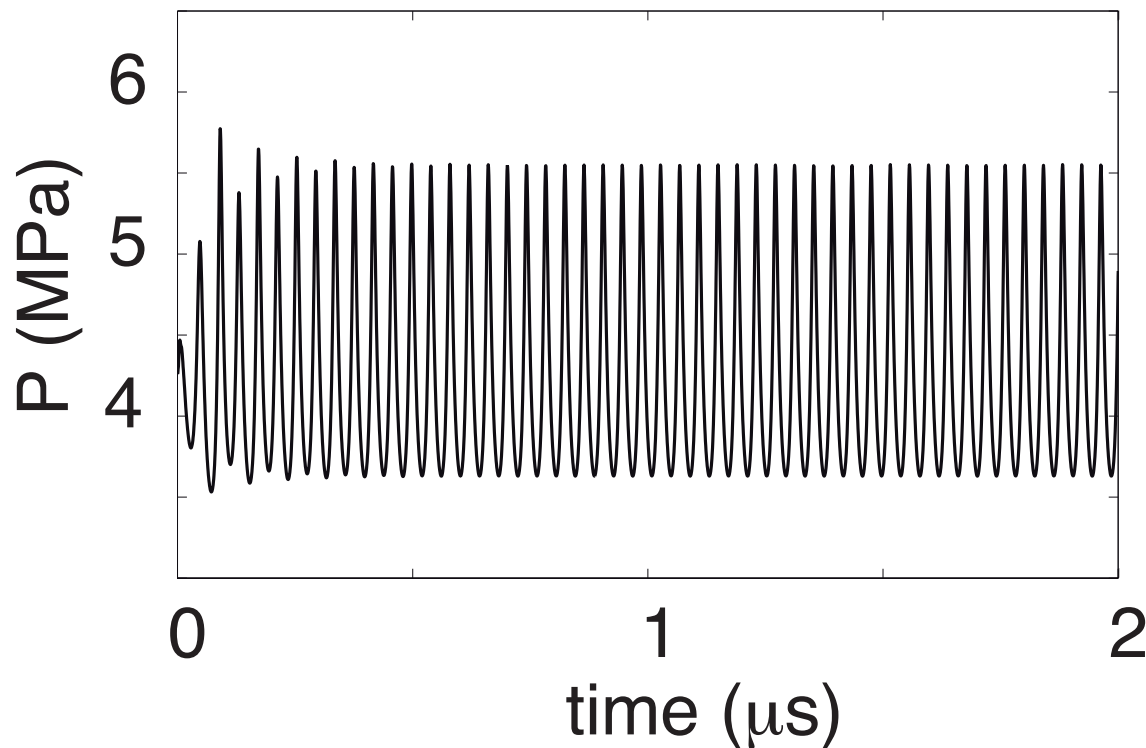


- Initialized with inviscid ZND solution.
- Moving frame travels at the CJ velocity.
- Integrated in time for long time behavior.

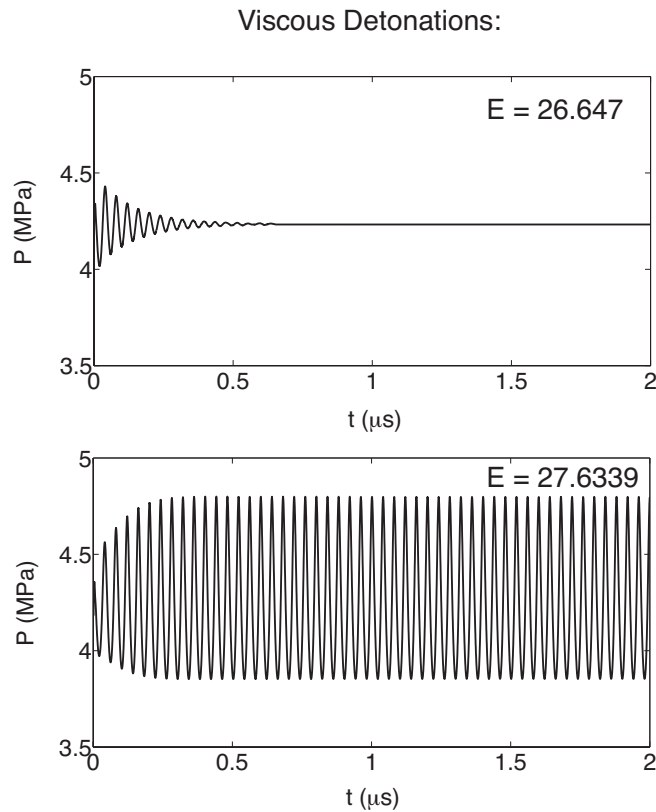
**Below a Critical Activation Energy, the Detonation is Stable**



**At Higher Activation Energy, Fundamental Harmonic  
Due to Balance Between Reaction and Advection  
Between Lead Shock and End of Reaction Zone:  
An Organ Pipe Resonance**



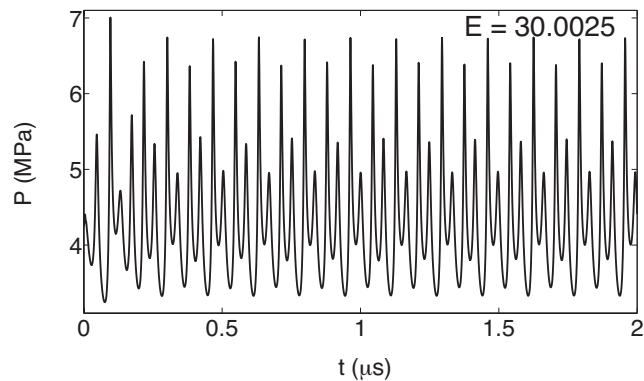
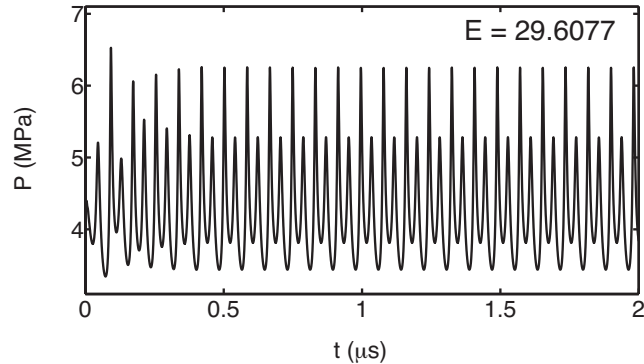
# Diffusion Delays Transition to Instability



- Lee and Stewart revealed for  $E < 25.26$  the steady ZND wave is linearly stable.
- For the inviscid case Henrick *et al.* found the stability limit at  $E_0 = 25.265 \pm 0.005$ .
- In the viscous case  $E = 26.647$  is still stable; however, above  $E_0 \approx 27.1404$  a period-1 limit cycle can be realized.

# Period-Doubling Phenomena Predicted

Viscous Detonations:



- As in the inviscid limit the viscous case goes through a period-doubling phase.
- For the inviscid case the period-doubling began at  $E_1 \approx 27.2$ .
- In the viscous case the beginning of this period doubling is delayed to  $E_1 \approx 29.3116$ .



## Diffusion Delays Transition to Chaos

- In the inviscid limit, the point where bifurcation points accumulate is found to be  $E_\infty \approx 27.8324$ .
- For the viscous case,  $L_\mu/L_{1/2} = 1/10$ , the accumulation point is delayed until  $E_\infty \approx 30.0411$ .
- For  $E > 30.0411$ , a region exists with many relative maxima in the detonation pressure; it is likely the system is in the chaotic regime.

## Approximations to Feigenbaum's Constant

$$\delta_\infty = \lim_{n \rightarrow \infty} \delta_n = \lim_{n \rightarrow \infty} \frac{E_n - E_{n-1}}{E_{n+1} - E_n}$$

Feigenbaum predicted  $\delta_\infty \approx 4.669201$ .

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	Inviscid	Inviscid	Viscous	Viscous
$n$	$E_n$	$\delta_n$	$E_n$	$\delta_n$
0	25.2650	-	27.1404	-
1	27.1875	3.86	29.3116	3.793
2	27.6850	4.26	29.8840	4.639
3	27.8017	4.66	30.0074	4.657
4	27.82675	-	30.0339	-

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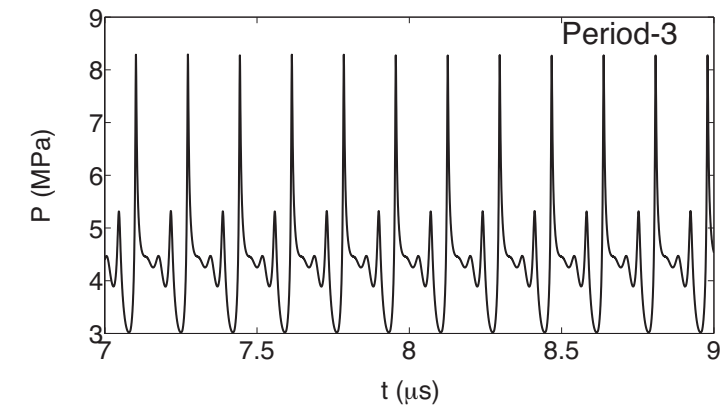
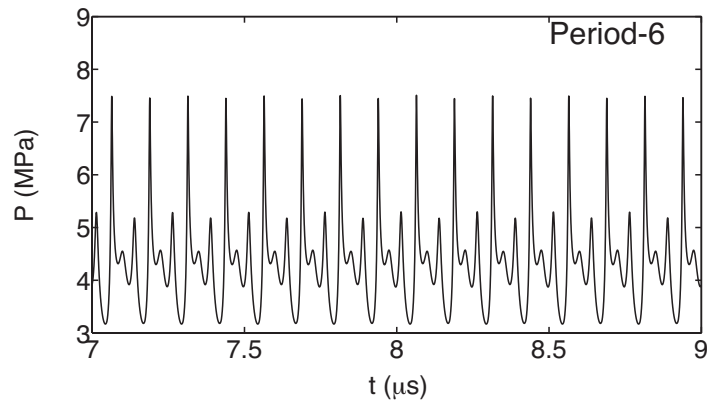
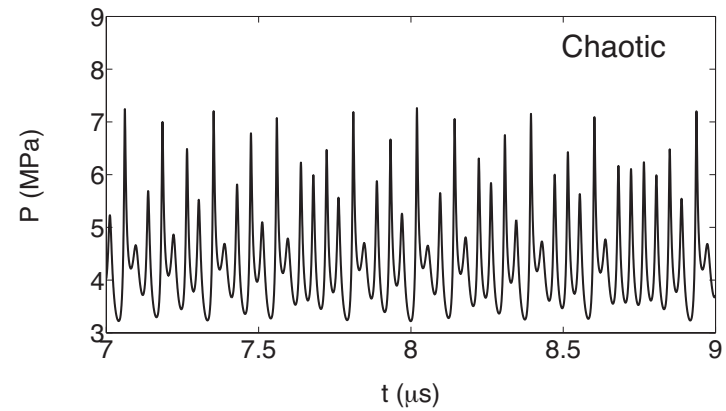
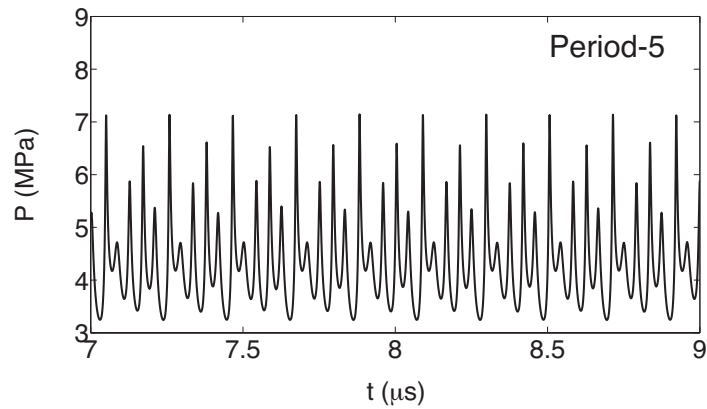
## Similar Behavior to Logistics Map:

$$x_{n+1} = rx_n(1 - x_n)$$

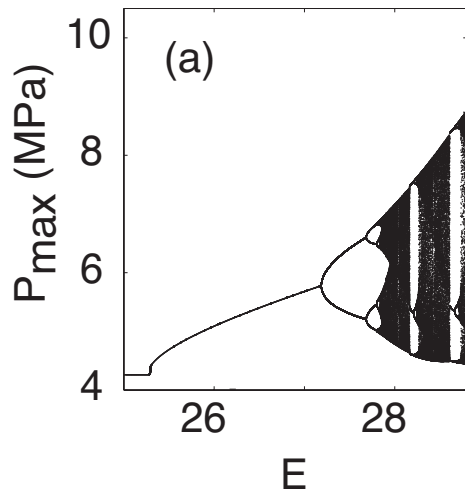
- The period-doubling behavior and transition to chaos predicted in both the viscous and inviscid limit have striking similarities to that of the logistic map.
- Within the chaotic region, there exist pockets of order.
- Periods of 5, 6, and 3 are found within this region.

# Chaos and Order

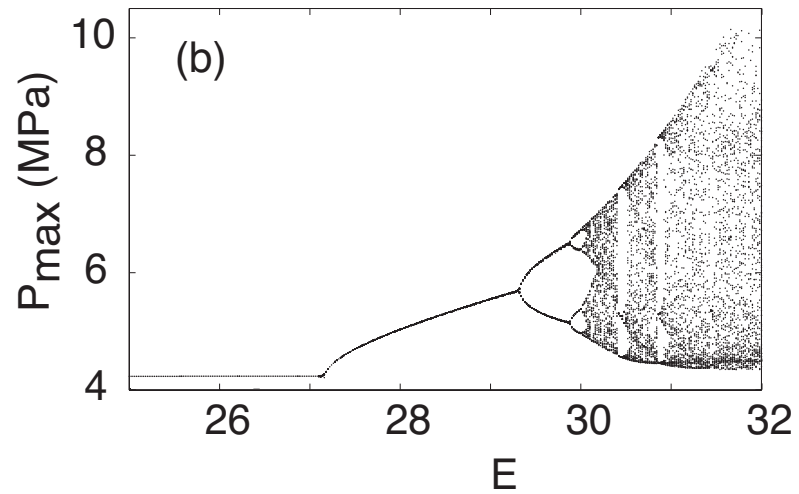
Viscous Detonations:



# Diffusion Delays Instability: Bifurcation Diagram

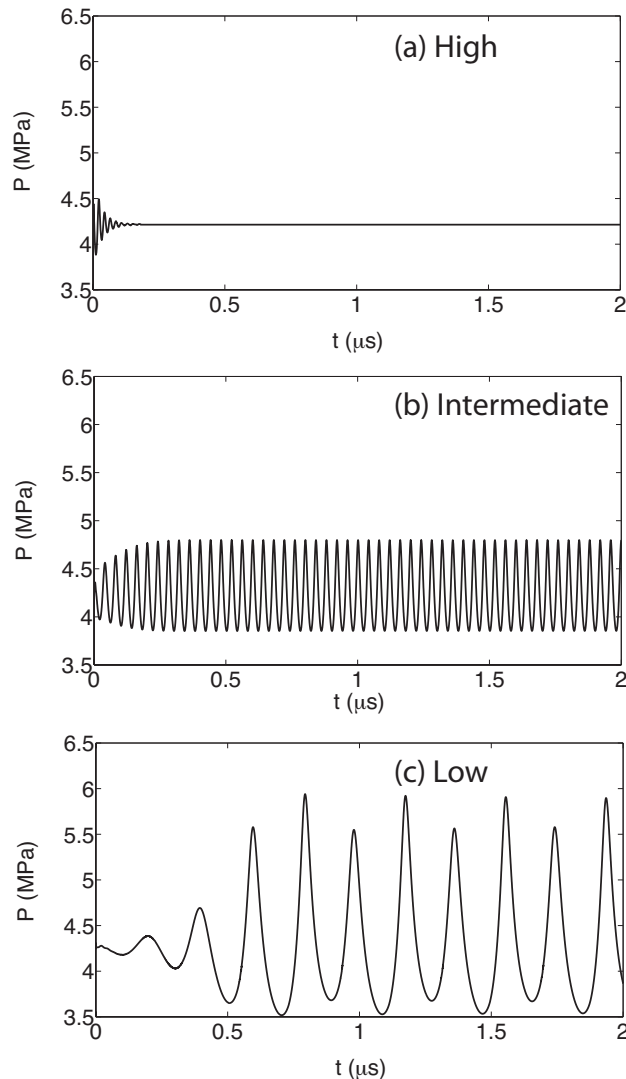


a) no diffusion



b) diffusion

# Diminishing Diffusion De-Stabilizes ( $E = 27.6339$ )



- The system undergoes transition from a stable detonation to a period-1 limit cycle, to a period-2 limit cycle.
- The amplitude of pulsations increases.
- The frequency decreases.

## Harmonic Analysis - PSD

- Harmonic analysis can be used to extract the multiple frequencies of a signal
- The discrete one-sided mean-squared amplitude Power Spectral Density (PSD) for the pressure was used

$$\Phi_d(0) = \frac{1}{N^2} |P_o|^2,$$

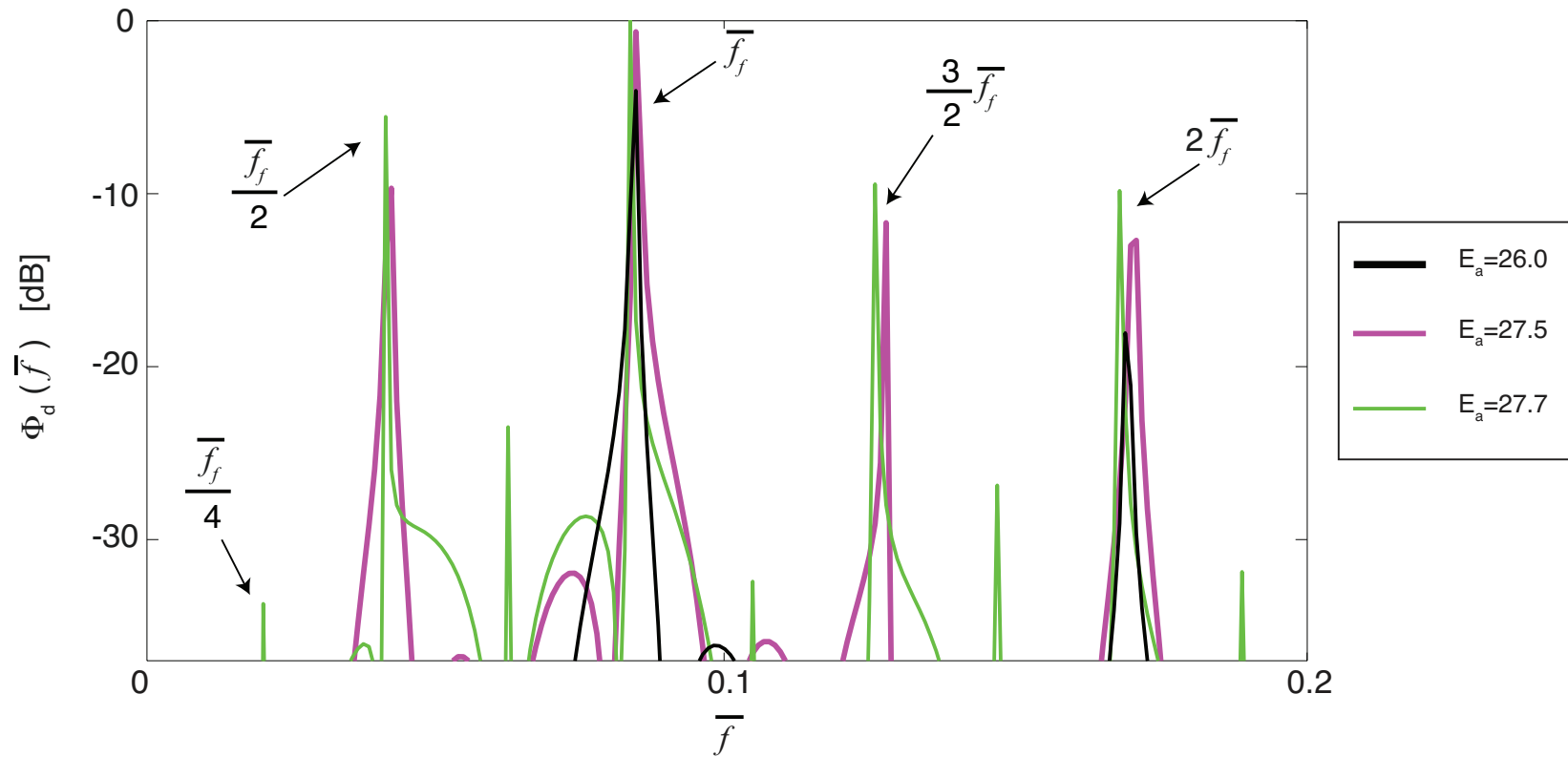
$$\Phi_d(\bar{f}_k) = \frac{2}{N^2} |P_k|^2, \quad k = 1, 2, \dots, (N/2 - 1),$$

$$\Phi_d(N/2) = \frac{1}{N^2} |P_{N/2}|^2,$$

where  $P_k$  is the standard discrete Fourier Transform of  $p$ ,

$$P_k = \sum_{n=0}^{N-1} p_n \exp\left(-\frac{2\pi i n k}{N}\right), \quad k = 0, 1, 2, \dots, N/2.$$

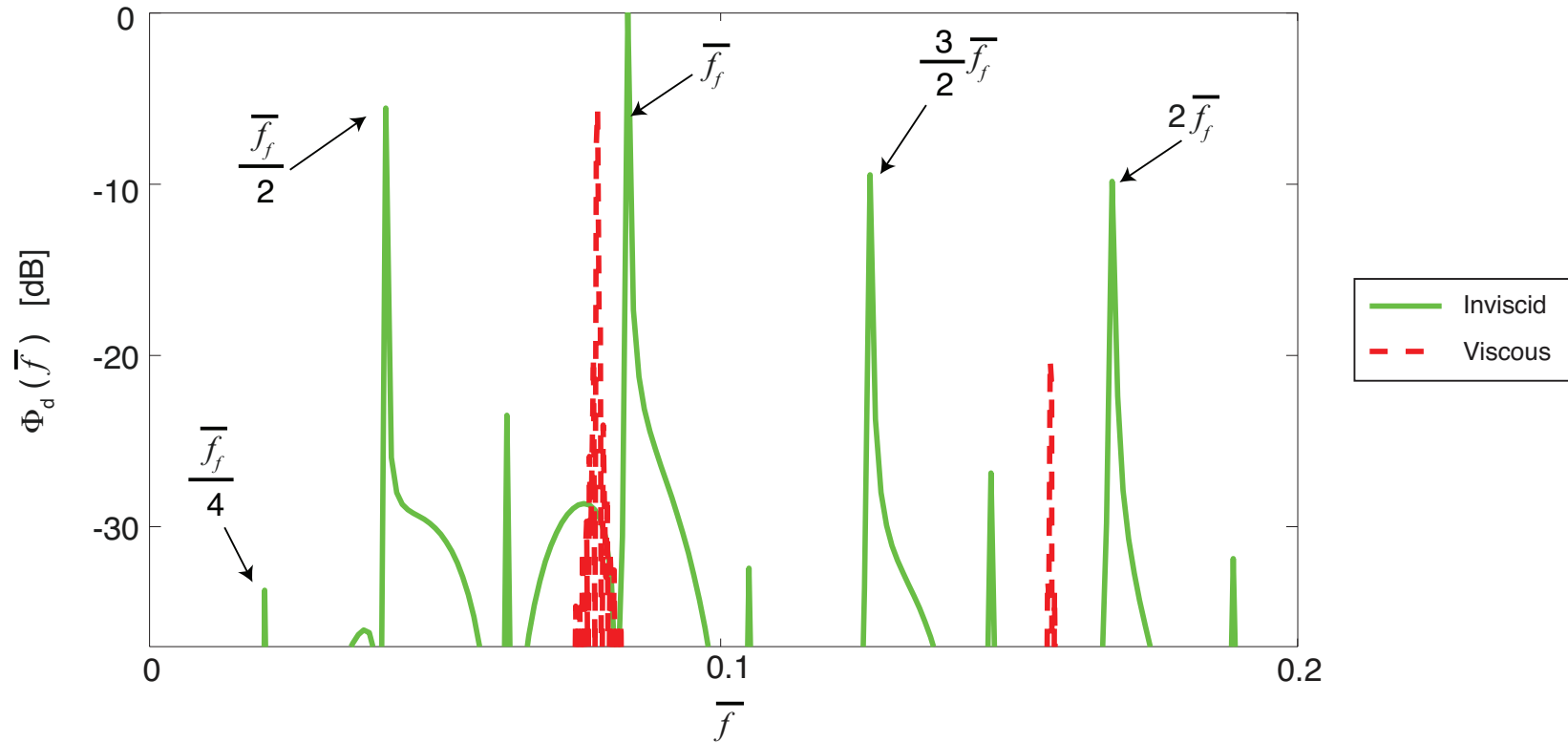
# Higher Order Harmonics Predicted as Activation Energy Increases





# Diffusion Modulates the Amplitude and Shifts the Frequency

$$E_a = 27.7$$



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## **Simple One-Step Model: Conclusions**

- Dynamics of one-dimensional detonations are influenced significantly by mass, momentum, energy diffusion in the region of instability.
  - In general, the effect of diffusion is stabilizing.
  - Bifurcation and transition to chaos show similarities to the logistic map.
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# Detailed Reaction Kinetics Model

# Unsteady, Compressible, Reactive Navier-Stokes Equations

$$\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{u}) = 0,$$

$$\frac{\partial}{\partial t} (\rho \mathbf{u}) + \nabla \cdot (\rho \mathbf{u} \mathbf{u} + p \mathbf{I} - \boldsymbol{\tau}) = \mathbf{0},$$

$$\frac{\partial}{\partial t} \left( \rho \left( e + \frac{\mathbf{u} \cdot \mathbf{u}}{2} \right) \right) + \nabla \cdot \left( \rho \mathbf{u} \left( e + \frac{\mathbf{u} \cdot \mathbf{u}}{2} \right) + (p \mathbf{I} - \boldsymbol{\tau}) \cdot \mathbf{u} + \mathbf{q} \right) = 0,$$

$$\frac{\partial}{\partial t} (\rho Y_i) + \nabla \cdot (\rho \mathbf{u} Y_i + \mathbf{j}_i) = \overline{M}_i \dot{\omega}_i,$$

$$p = \mathcal{R} T \sum_{i=1}^N \frac{Y_i}{M_i}, \quad e = e(T, Y_i), \quad \dot{\omega}_i = \dot{\omega}_i(T, Y_i),$$

$$\mathbf{j}_i = \rho \sum_{\substack{k=1 \\ k \neq i}}^N \frac{\overline{M}_i D_{ik} Y_k}{\overline{M}} \left( \frac{\nabla y_k}{y_k} + \left( 1 - \frac{\overline{M}_k}{\overline{M}} \right) \frac{\nabla p}{p} \right) - \frac{D_i^T \nabla T}{T},$$

$$\boldsymbol{\tau} = \mu \left( \nabla \mathbf{u} + (\nabla \mathbf{u})^T - \frac{2}{3} (\nabla \cdot \mathbf{u}) \mathbf{I} \right),$$

$$\mathbf{q} = -k \nabla T + \sum_{i=1}^N \mathbf{j}_i h_i - \mathcal{R} T \sum_{i=1}^N \frac{D_i^T}{M_i} \left( \frac{\nabla \bar{y}_i}{\bar{y}_i} + \left( 1 - \frac{\overline{M}_i}{\overline{M}} \right) \frac{\nabla p}{p} \right).$$

# Computational Methods

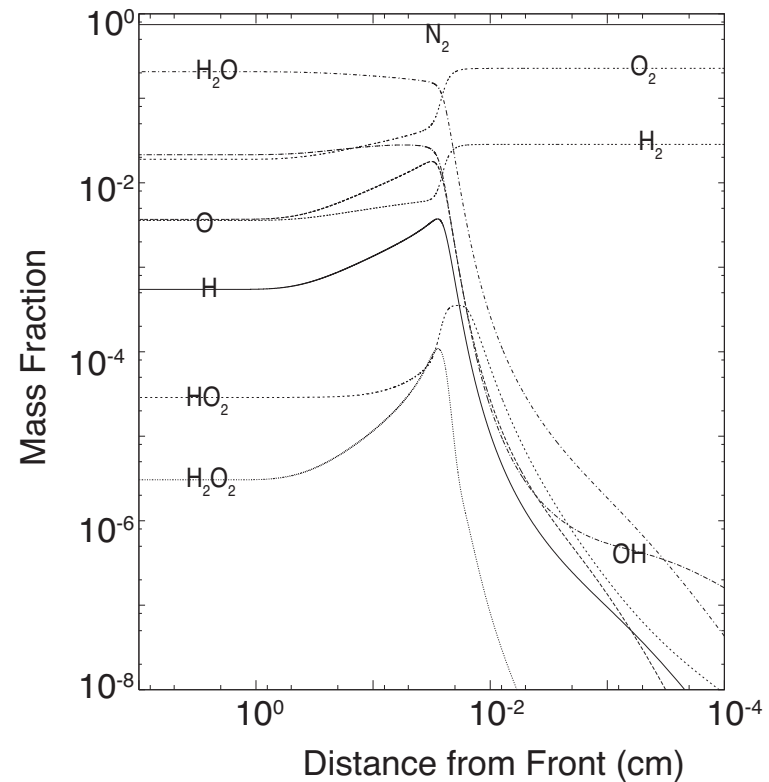
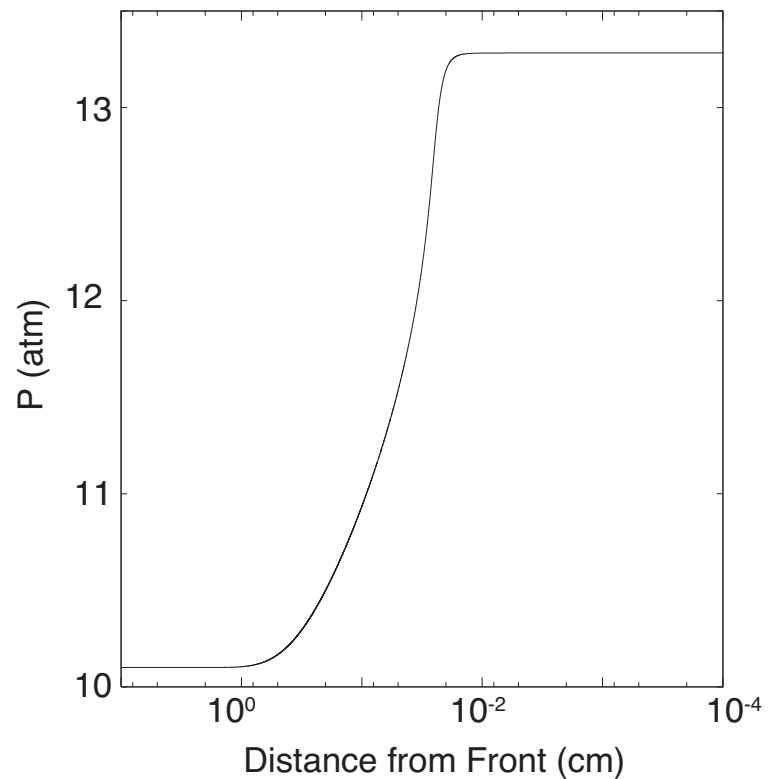
- Inviscid
  - Shock-fitting : Fifth order algorithm adapted from Henrick *et al.*, *JCP*.
  - Shock-capturing : Second order min-mod algorithm
- Viscous
  - Wavelet method (WAMR), developed by Vasilyev and Paolucci, *JCP*
  - User-defined threshold parameter  $\epsilon$  controls error: *automatic verification!*

$$u^J(\mathbf{x}) = \underbrace{\sum_{\mathbf{k}} u_{0,\mathbf{k}} \Phi_{0,\mathbf{k}}(\mathbf{x}) + \sum_{j=0}^{J-1} \sum_{\{\lambda: |d_{j,\lambda}| \geq \epsilon\}} d_{j,\lambda} \Psi_j(\mathbf{x})}_{u_\epsilon^J} + \underbrace{\sum_{j=0}^{J-1} \sum_{\{\lambda: |d_{j,\lambda}| < \epsilon\}} d_{j,\lambda} \Psi_j(\mathbf{x})}_{R_\epsilon^J}$$

- All methods used a fifth order explicit Runge-Kutta scheme for time integration

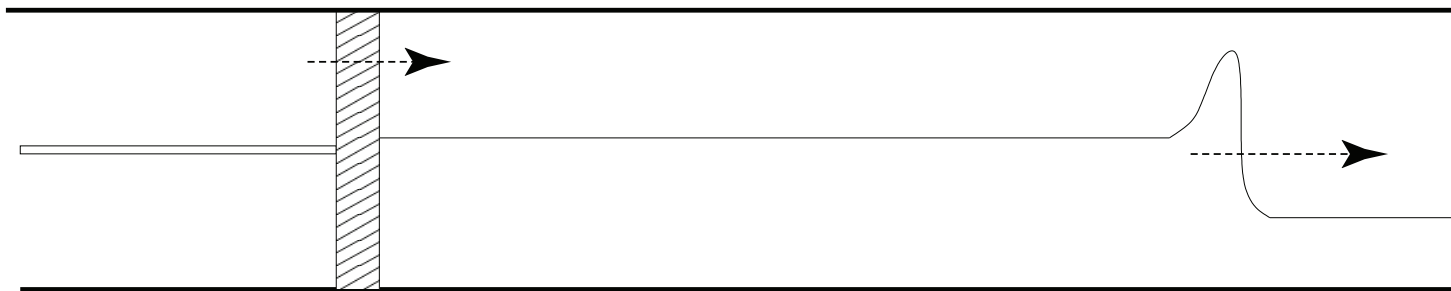
# Typical Stable Steady Wave Profile

$$f = 1.15$$



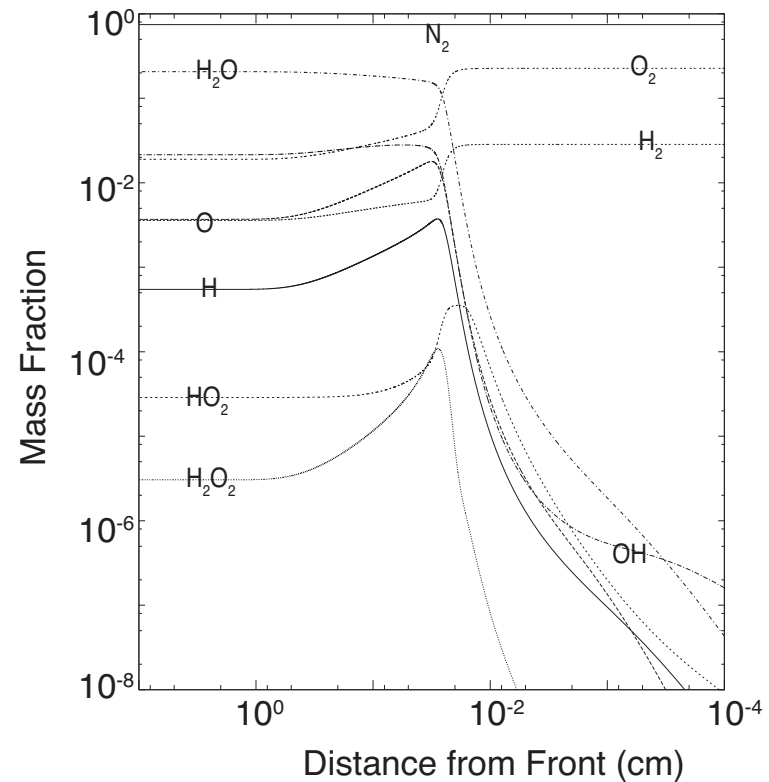
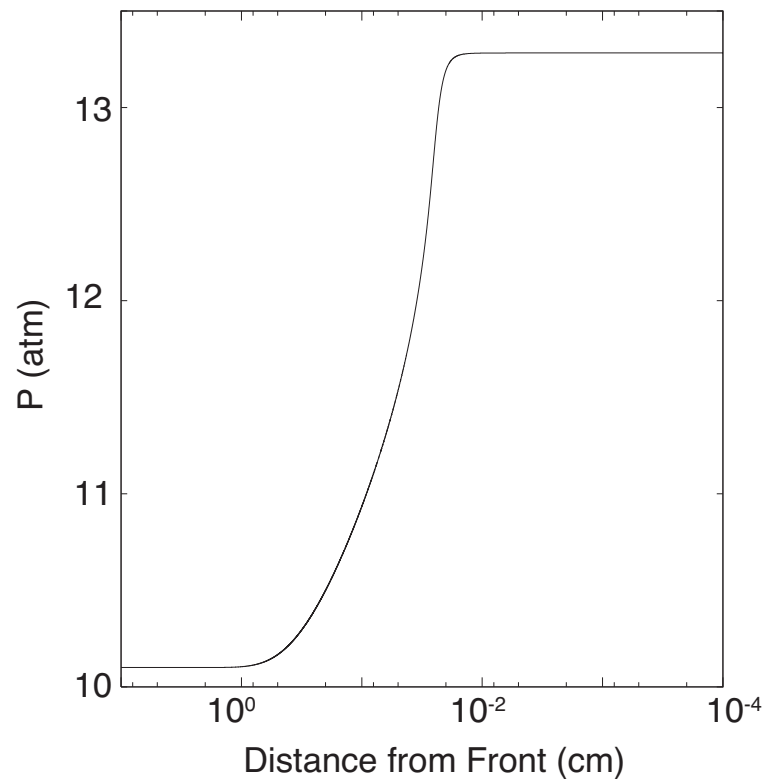
## Cases Examined

- Overdriven detonations with ambient conditions of 0.421 atm and 293.15 K
- Initial stoichiometric mixture of  $2\text{H}_2 + \text{O}_2 + 3.76\text{N}_2$
- $D_{CJ} \sim 1972$  m/s
- Overdrive is defined as  $f = D_o^2 / D_{CJ}^2$
- Overdrives of  $1.018 < f < 1.150$  were examined



# Typical Stable Steady Wave Profile

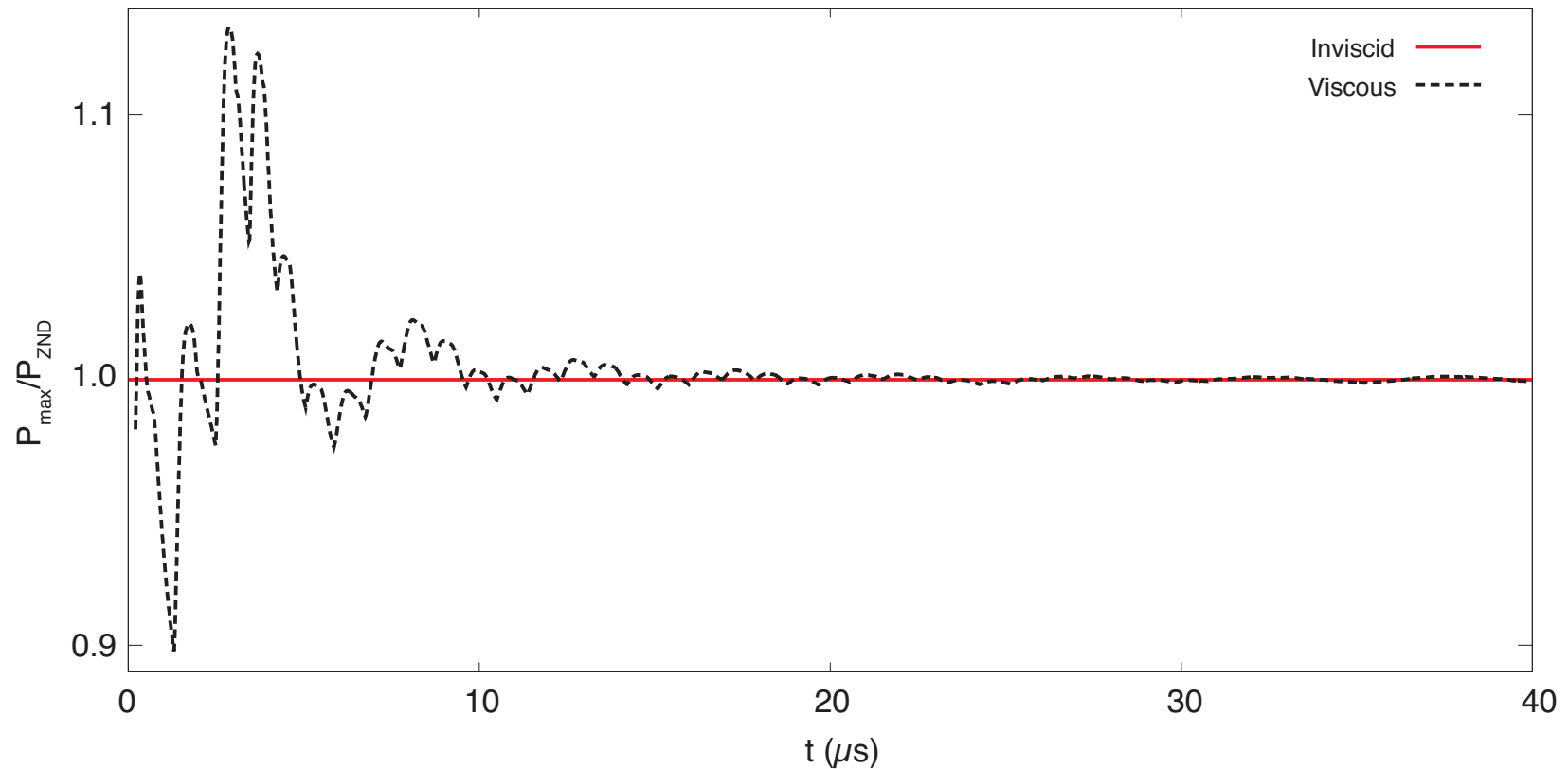
$$f = 1.15$$





# Stable Detonation at High Overdrive

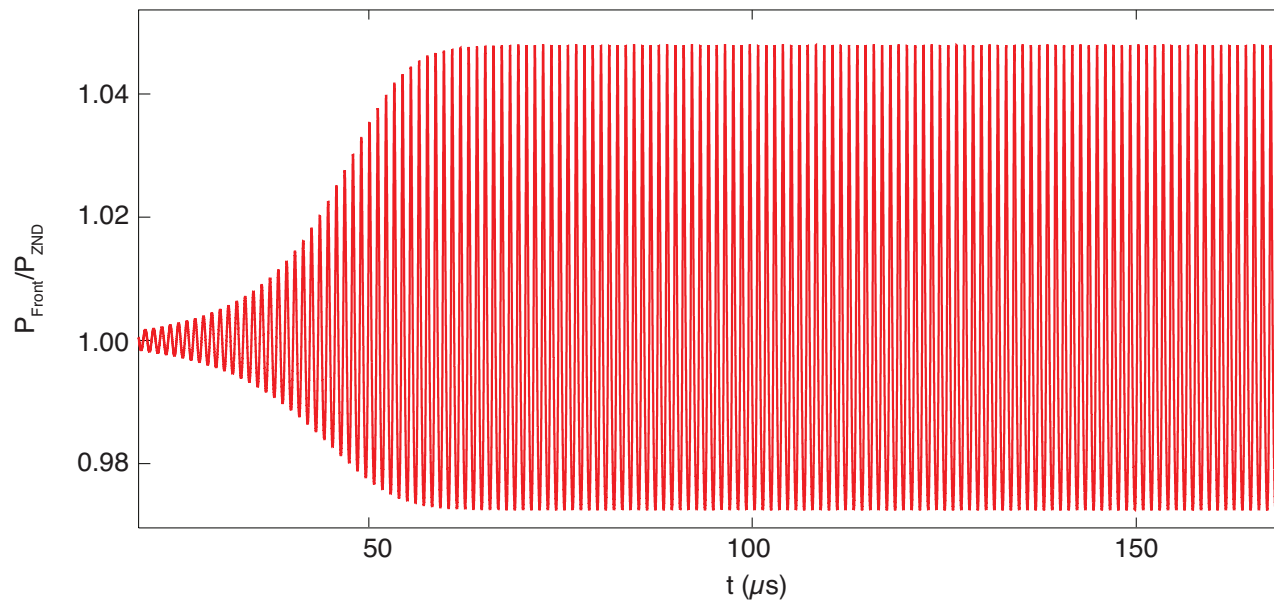
$$f = 1.15$$



For high enough overdrives, the detonation relaxes to a steady propagating wave in the inviscid case as well as in the diffusive case.

## Lower Overdrive: High Frequency Instability, No Diffusion

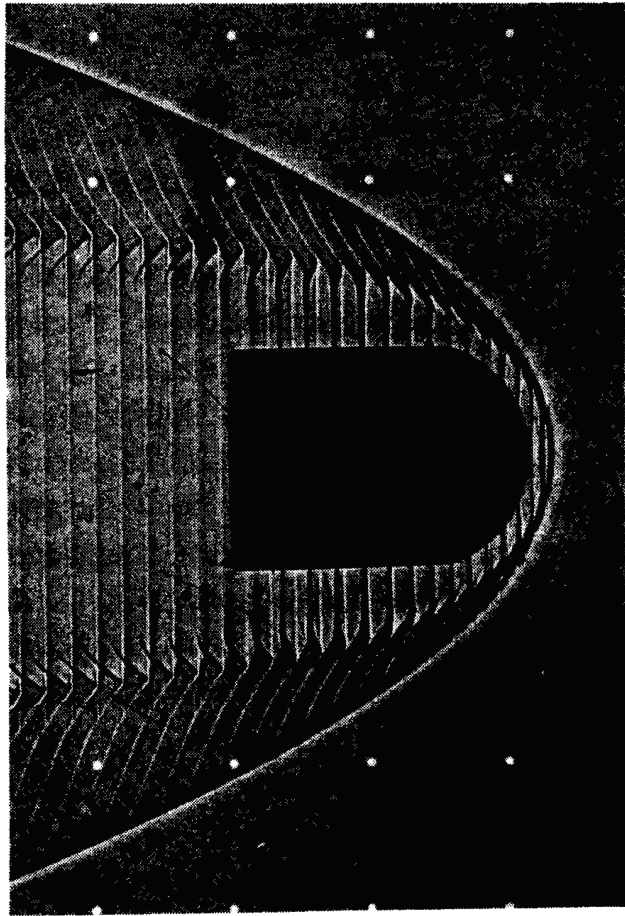
$$f = 1.10$$



A single fundamental frequency oscillation occurs at a frequency of  $0.97 \text{ MHz}$ .  
This frequency agrees with the experimental observations of Lehr (*Astro. Acta*, 1972).

*Organ pipe oscillation between shock and end of reaction zone:  $\nu \simeq a/\ell =$   
(1000 m/s)/(0.0001 m)  $\simeq$  10 MHz.*

## Lehr's High Frequency Instability

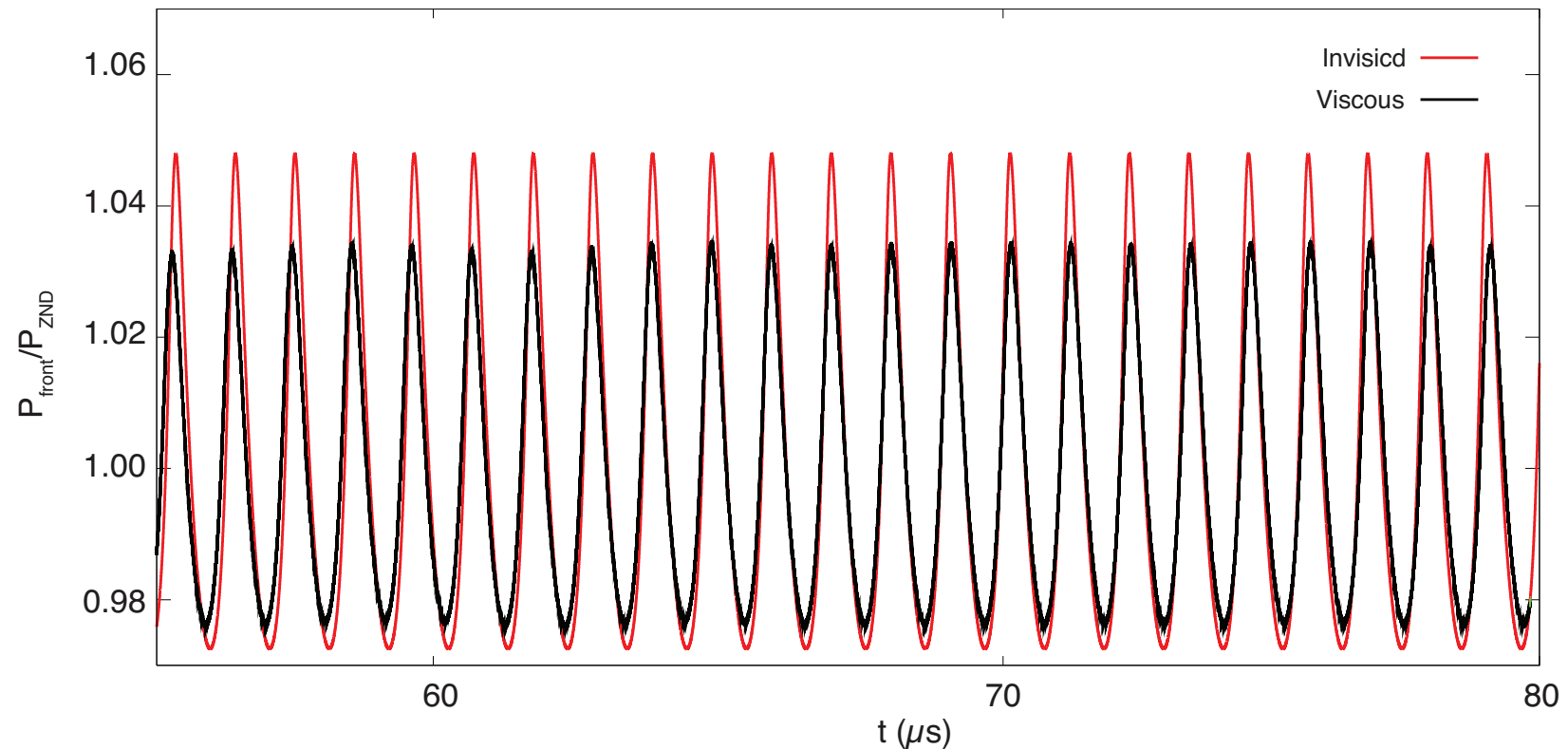


(*Astro. Acta*, 1972)

- Shock-induced combustion experiment (*Astro. Acta*, 1972)
- Stoichiometric mixture of  $2\text{H}_2 + \text{O}_2 + 3.76\text{N}_2$  at 0.421 atm
- Observed 1.04 MHz frequency for projectile velocity corresponding to  $f \approx 1.10$
- For  $f = 1.10$ , the predicted frequency of 0.97 MHz agrees with observed frequency and the prediction by Yungster and Radhakrishan of 1.06 MHz

# High Frequency Mode - Viscous vs. Inviscid

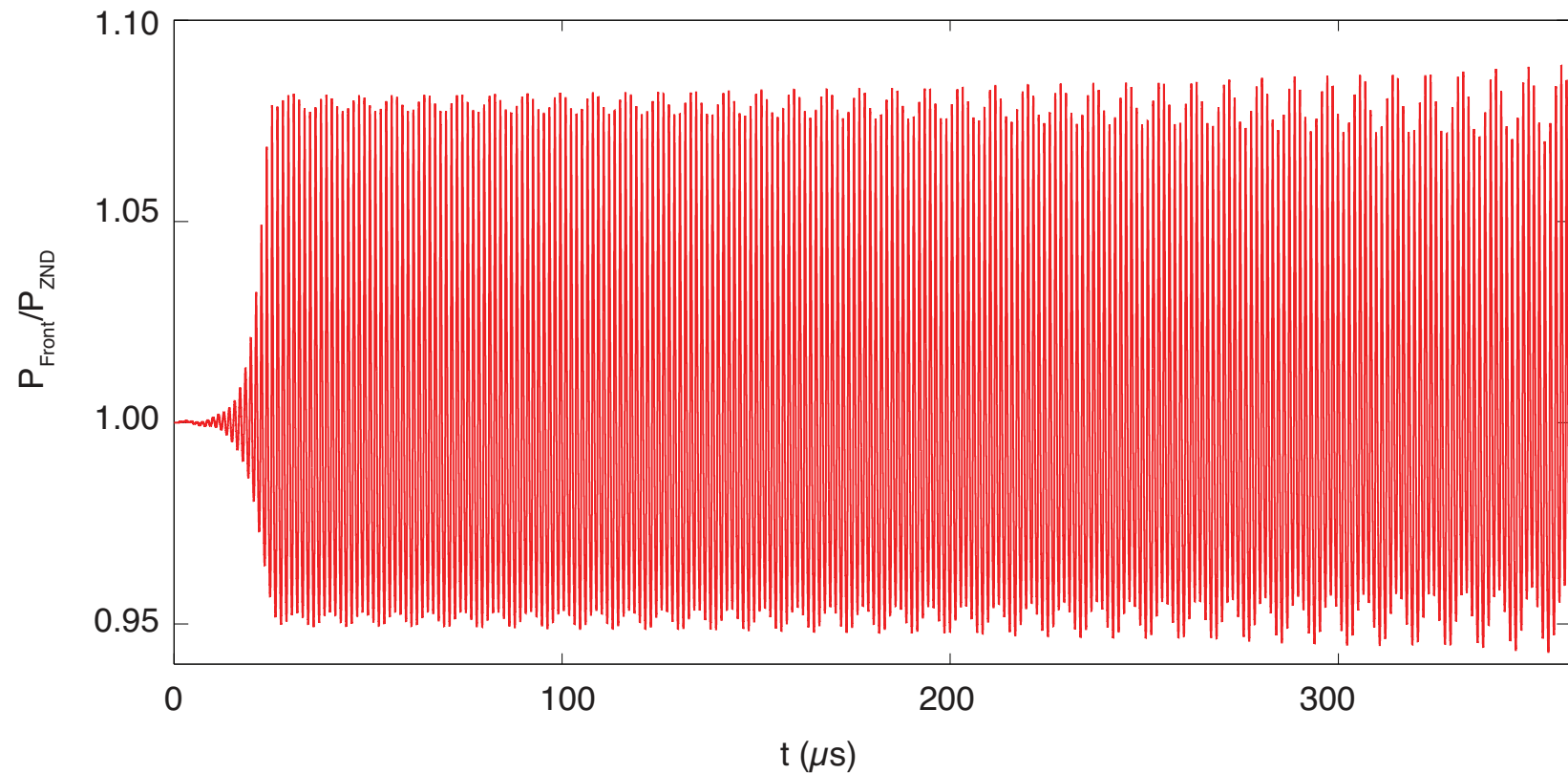
$$f = 1.10$$



The addition of viscosity has a stabilizing effect, decreasing the amplitude of the oscillations. The pulsation frequency relaxes to 0.97 MHz.

# Low Frequency Mode Appearance - Inviscid

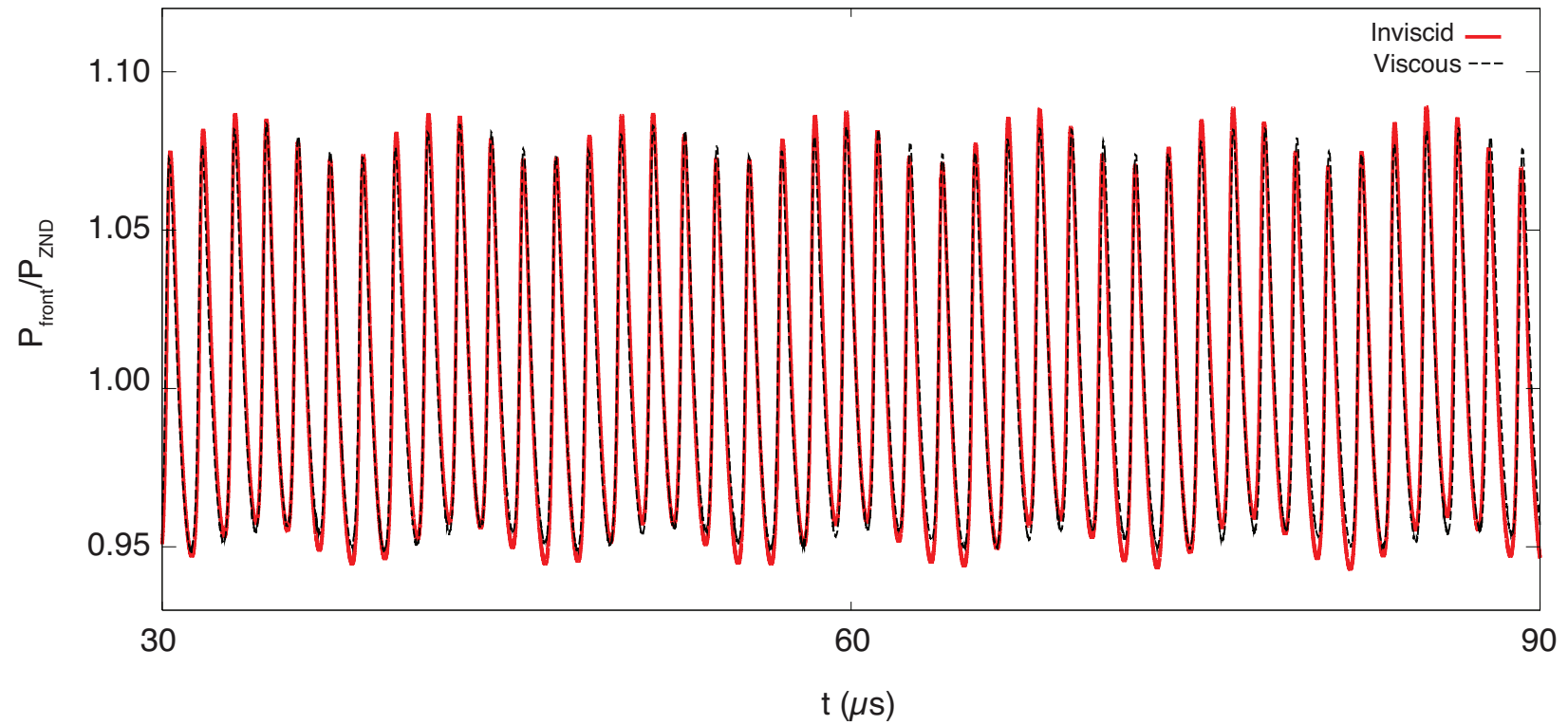
$$f = 1.035$$



As the overdrive is lowered, multiple frequencies appear, and the amplitude of the oscillations continues to grow. These multiple frequencies persist at long time.

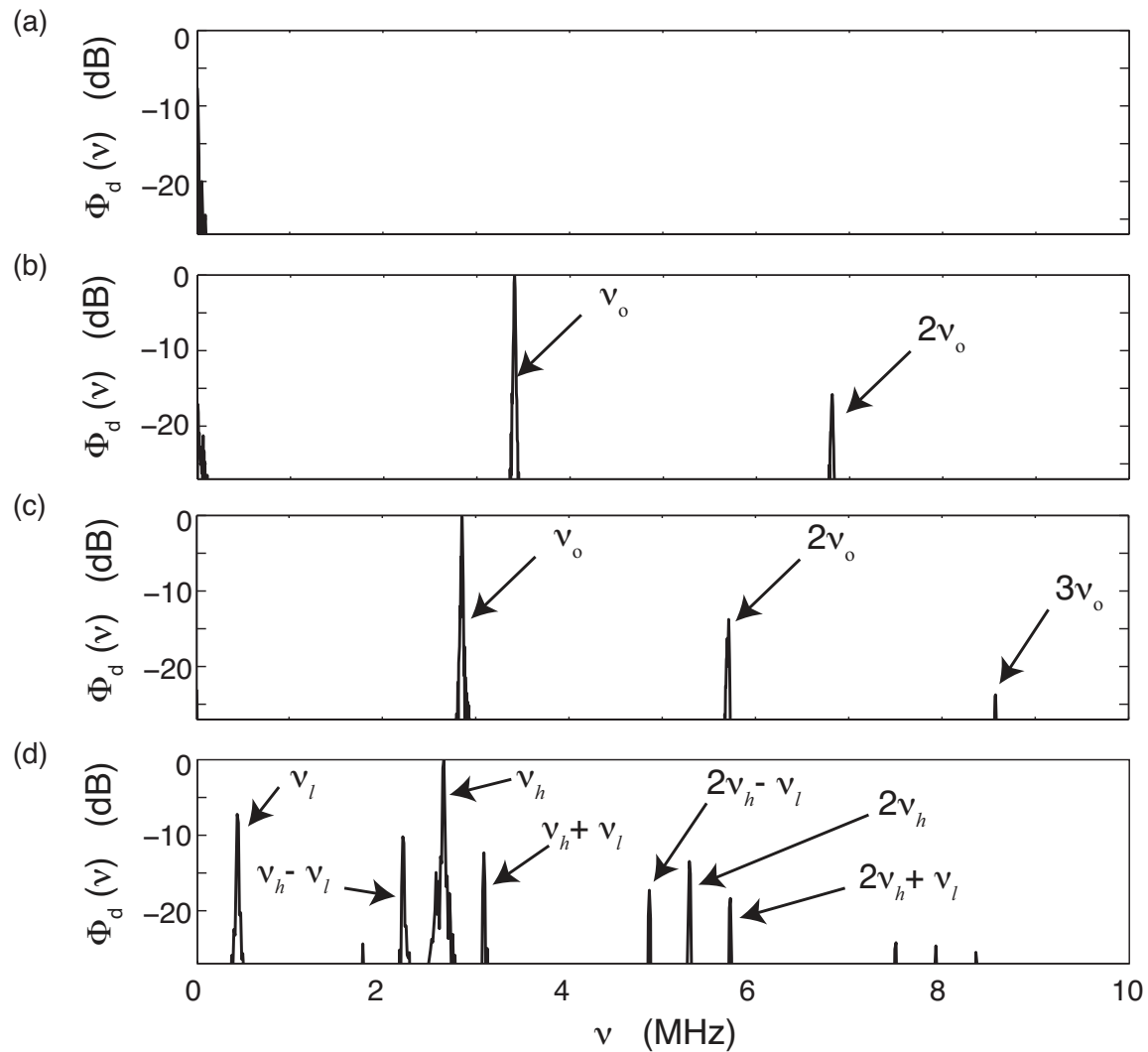
# Low Frequency Mode Appearance - Viscous vs. Inviscid

$$f = 1.035$$

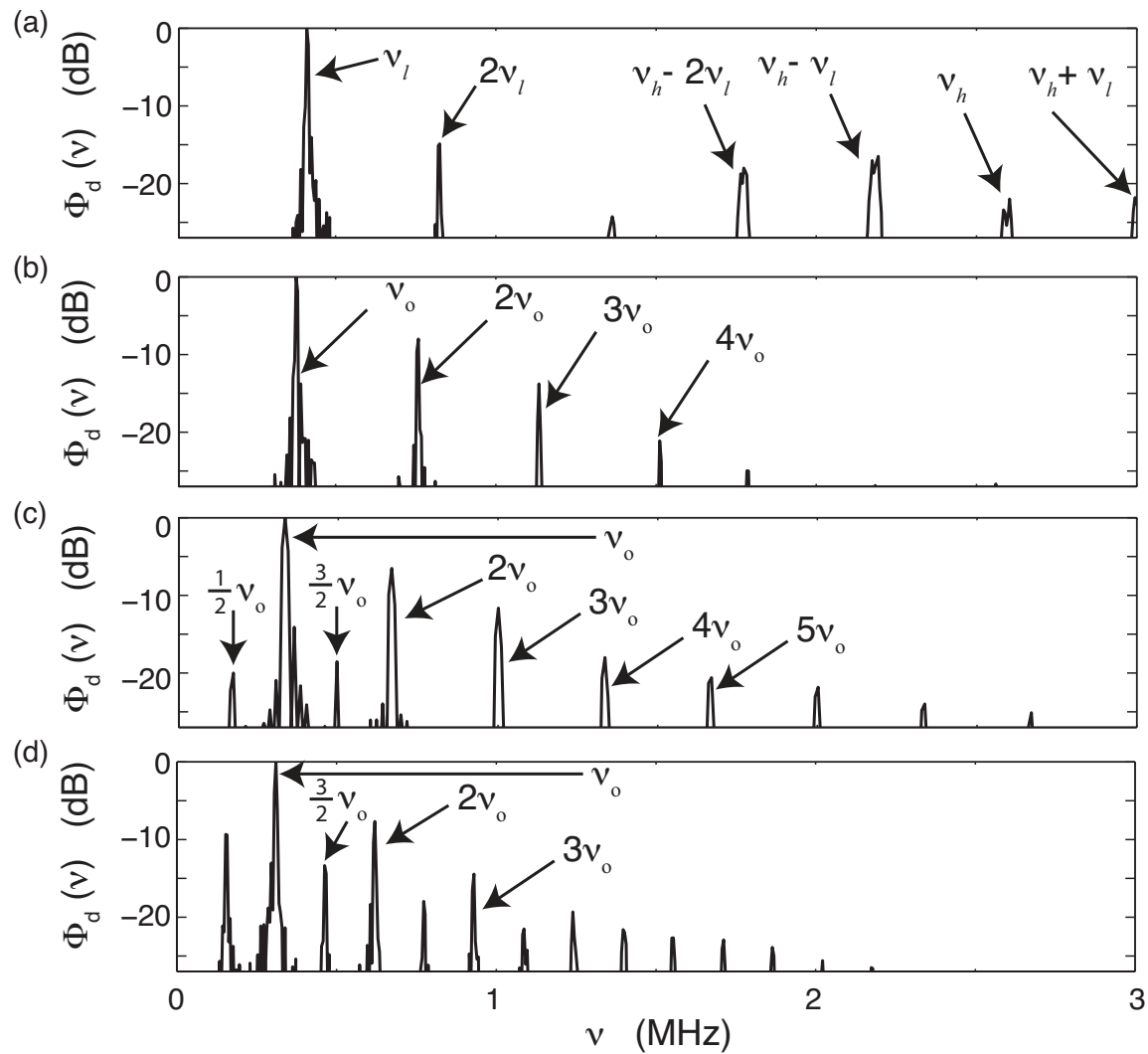


Viscosity still decreases the amplitude of oscillation, though the effect is reduced compared to higher overdrives. Longer times need further investigation.

# Viscous H<sub>2</sub>-Air Harmonics: Effect of Overdrive

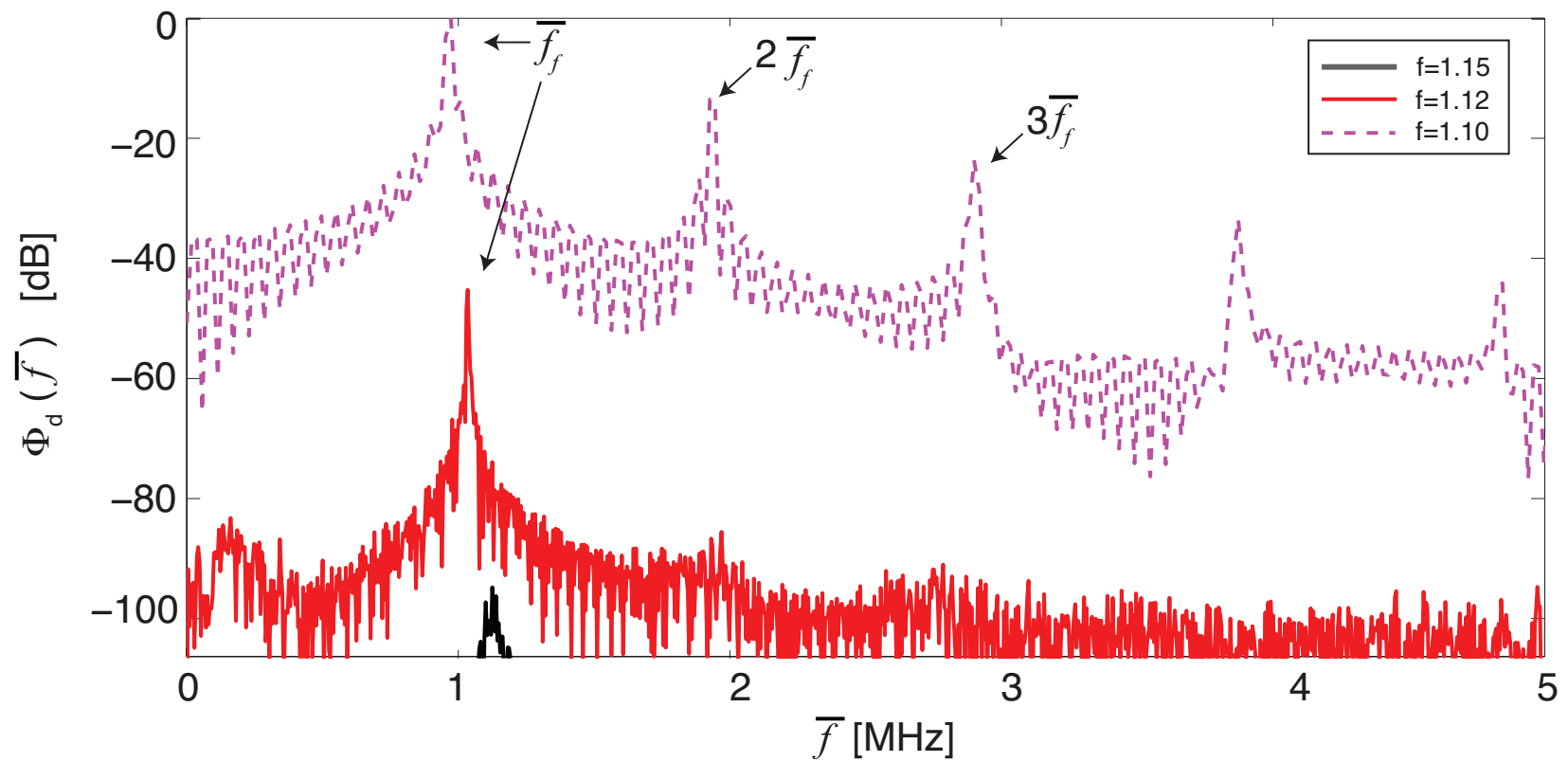


# Viscous H<sub>2</sub>-Air Harmonics: Effect of Overdrive

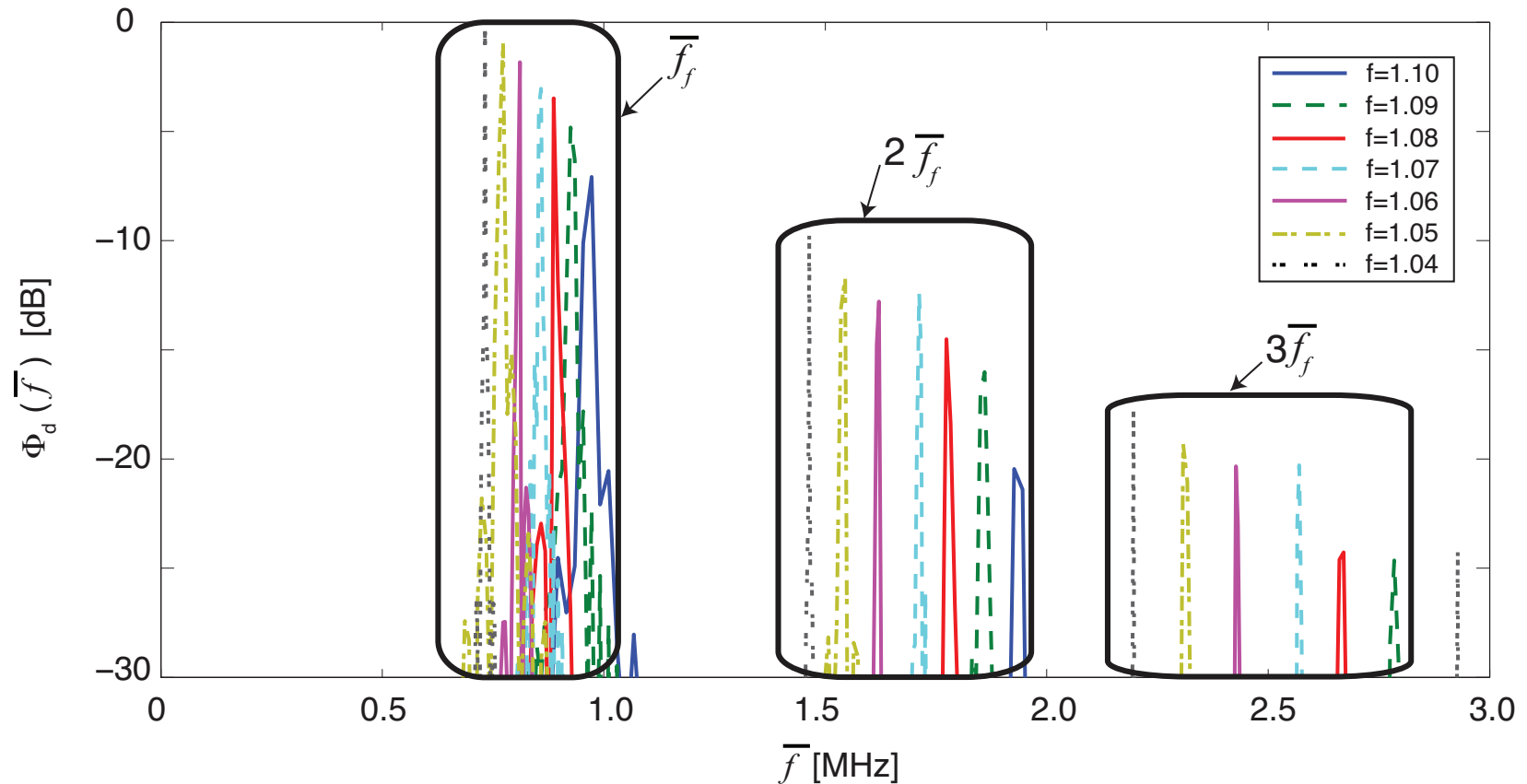




# H<sub>2</sub>-Air: Near Neutral Stability



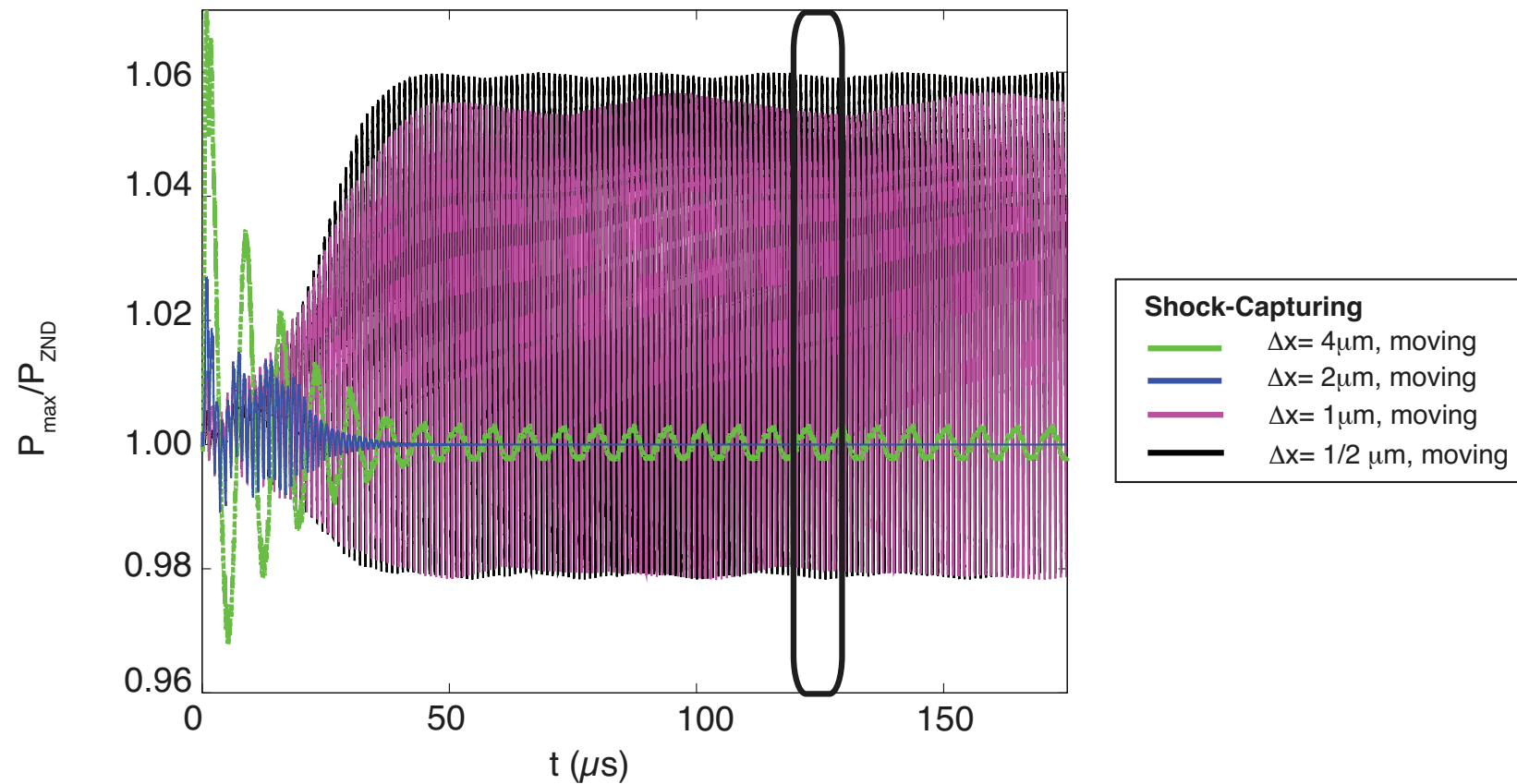
## H<sub>2</sub>-Air: High Frequency Shift



The amplitude of the oscillations continues to grow as the overdrive is lowered. There appears to be a near power-law decay in the amount of energy carried by the higher harmonics.

# Fine Grids Required for Accurate Shock-Capturing

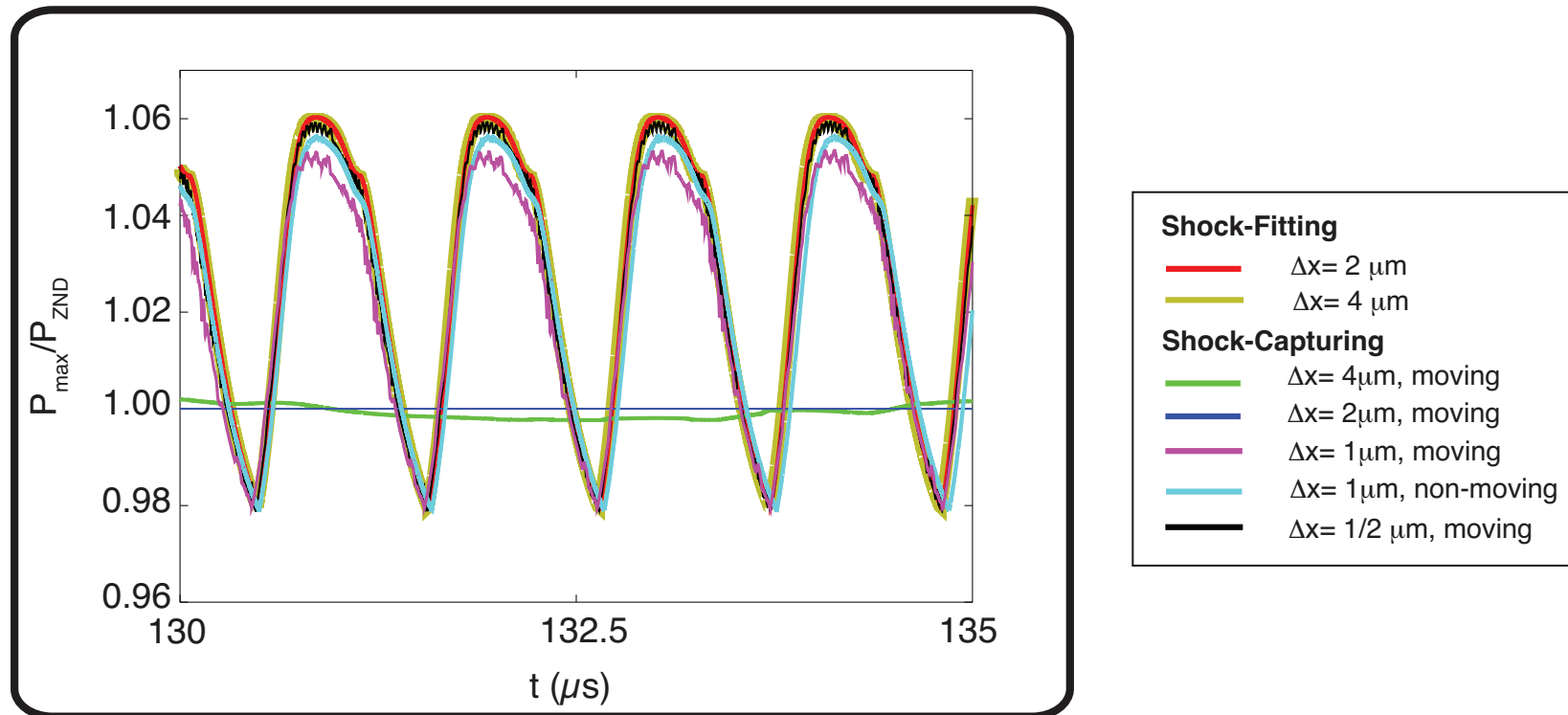
$$f = 1.10$$



Using the same grid size as shock-fitting ( $\Delta x = 4 \mu\text{m}$ ), shock-capturing misses the essential dynamics.

# Fine Grids Required for Accurate Shock-Capturing

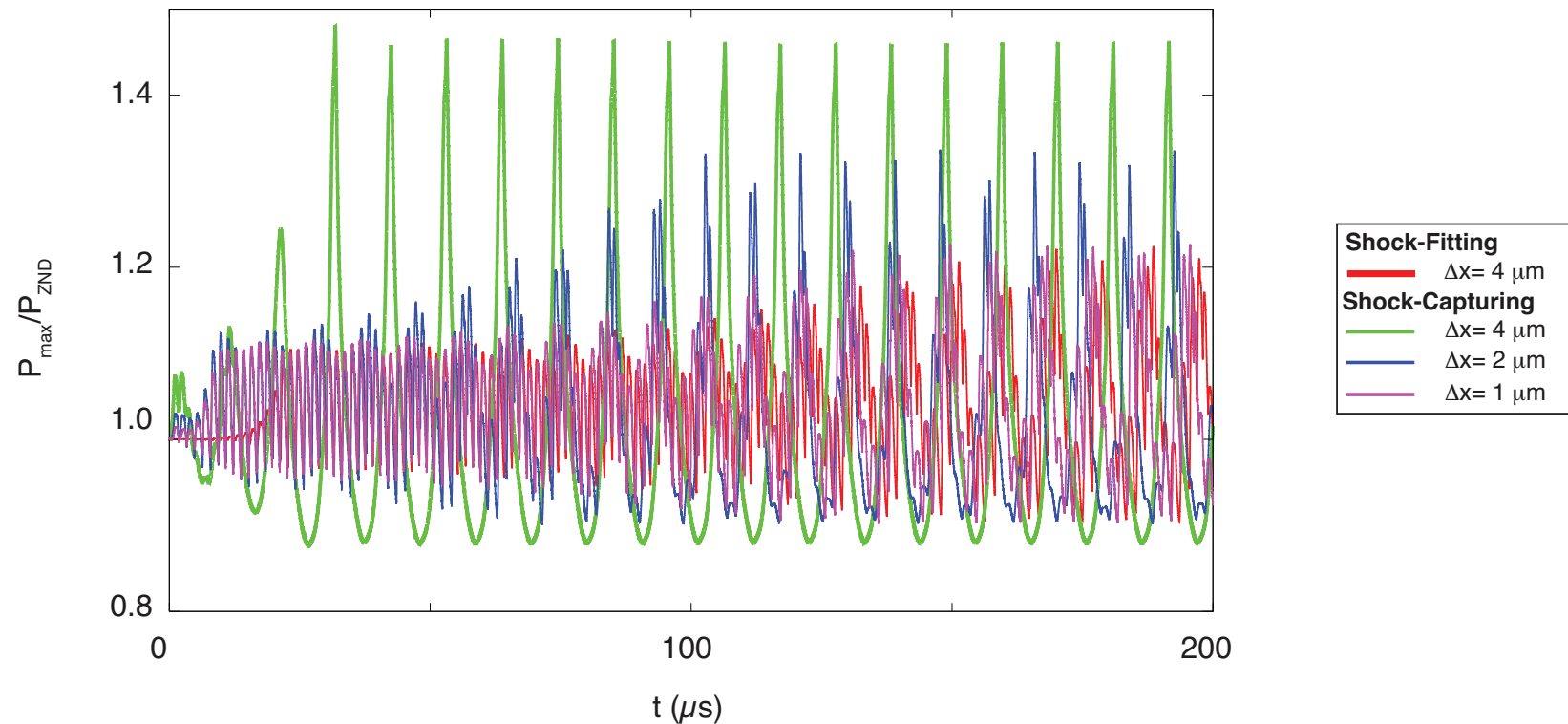
$$f = 1.10$$



Using a four times finer grid with shock-capturing than shock-fitting allows the pulsations to be captured. However, both much higher and lower frequency spurious oscillations are predicted as well.

# Fine Grids Required for Accurate Shock-Capturing

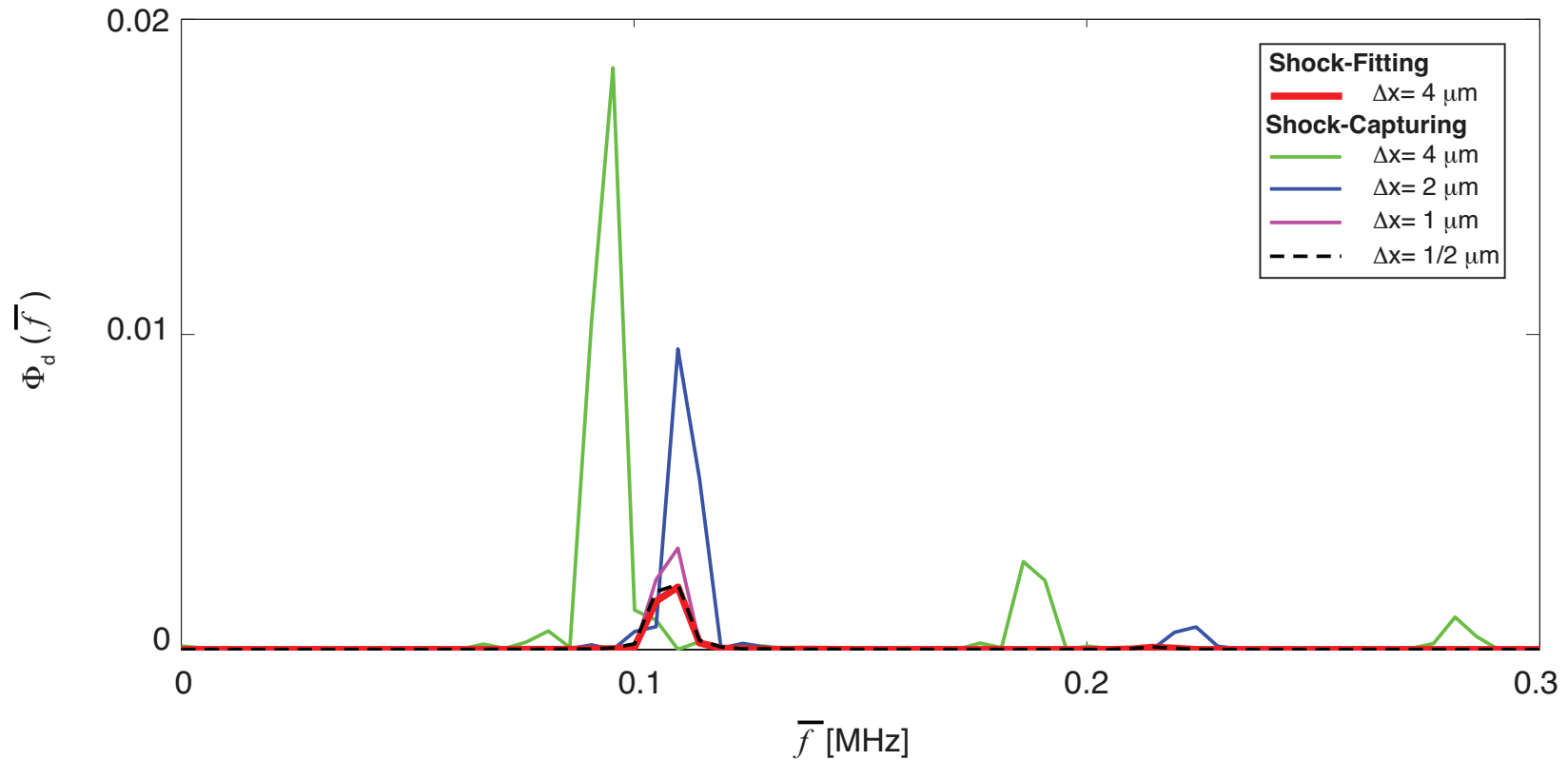
$$f = 1.023$$



Using the same grid size ( $\Delta x = 4 \mu\text{m}$ ) as shock-fitting, shock-capturing dramatically over-predicts the pulsation amplitude. In shock-capturing, a resolution of  $\Delta x = 1 \mu\text{m}$  is needed to begin capturing the essential dynamics at long time.

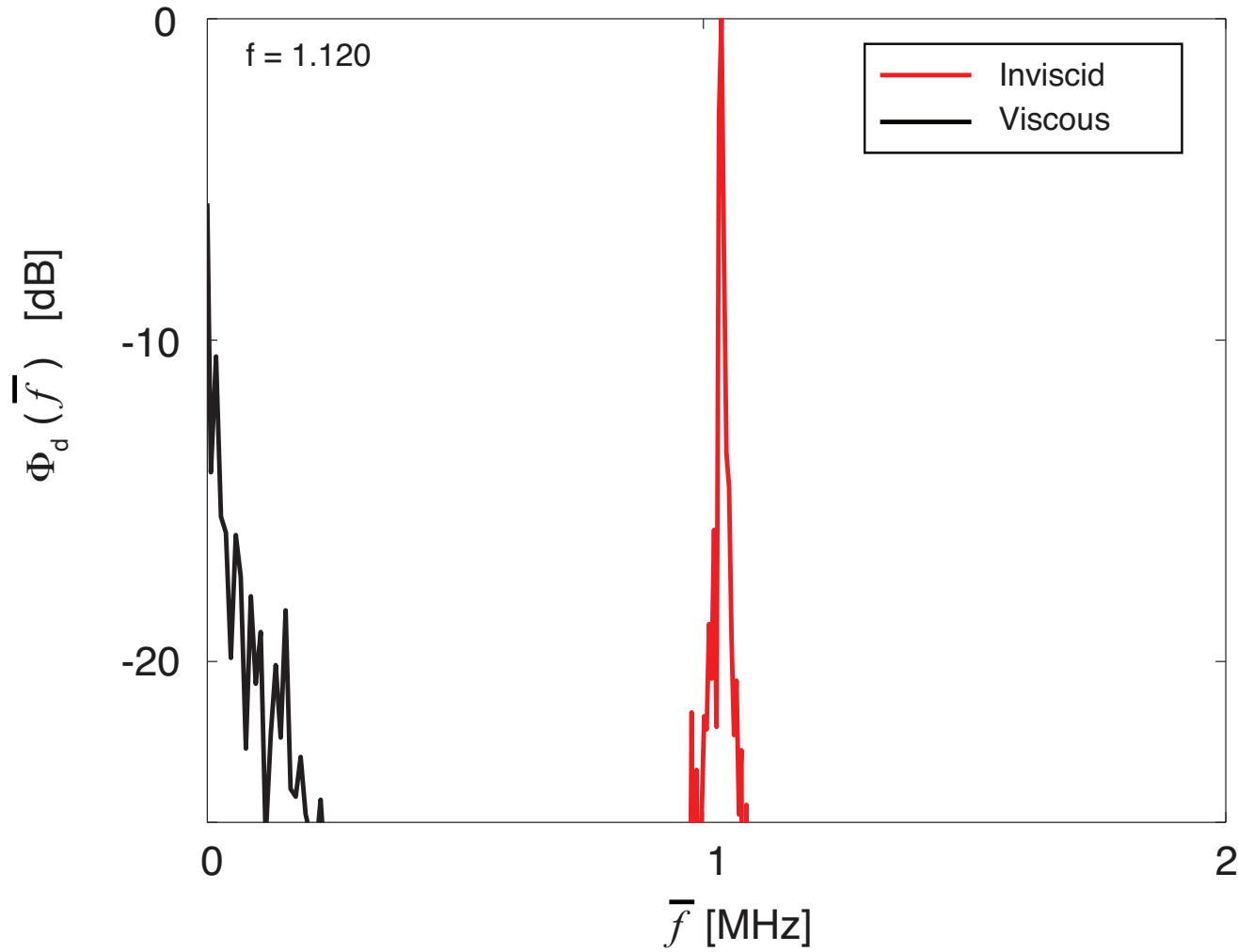
# Fine Grids Required for Accurate Shock-Capturing

$$f = 1.023$$

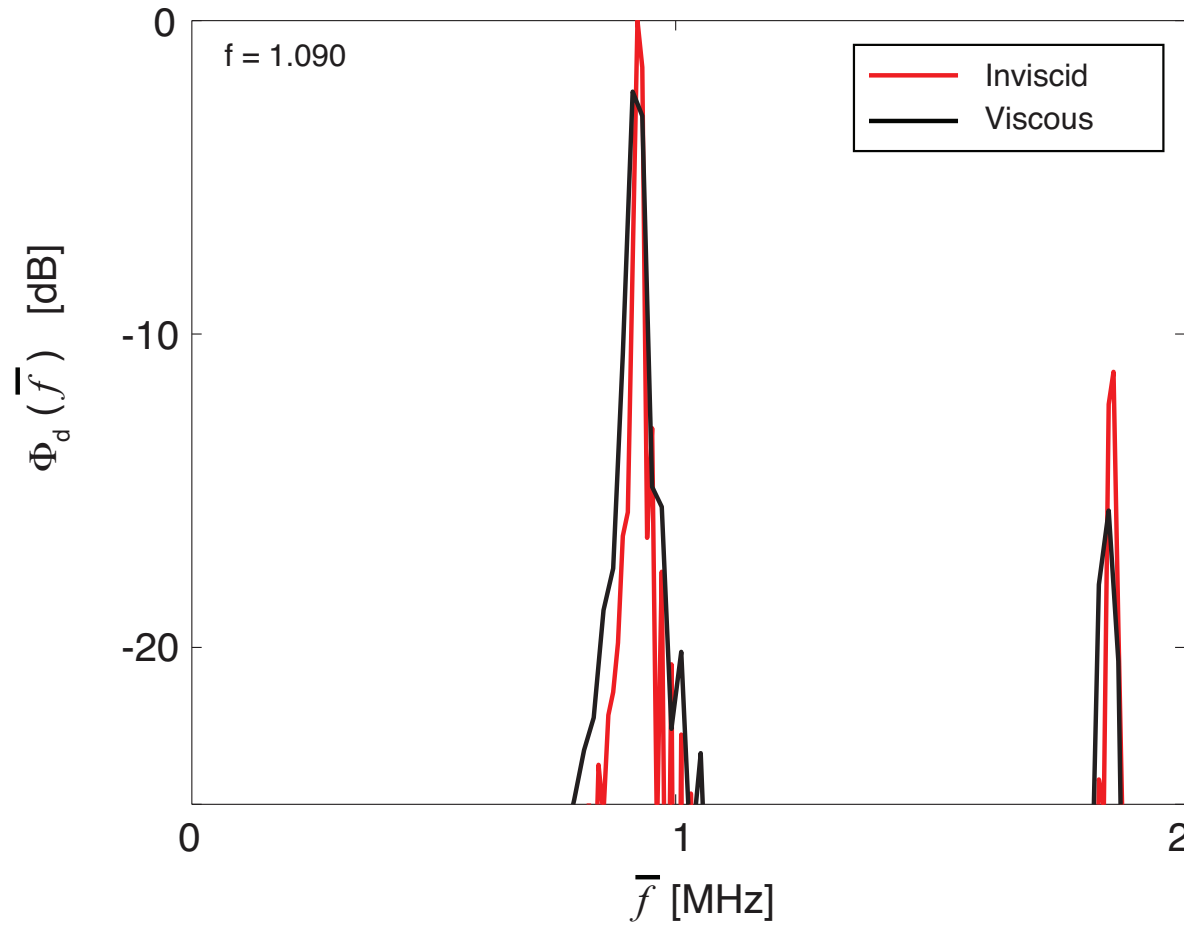


Only when  $\Delta x = 1/2 \mu\text{m}$  is used does the PSD of shock-capturing become nearly indistinguishable with that of shock-fitting.

# Near the Neutral Stability Boundary, Diffusion Damps the Small Oscillations.

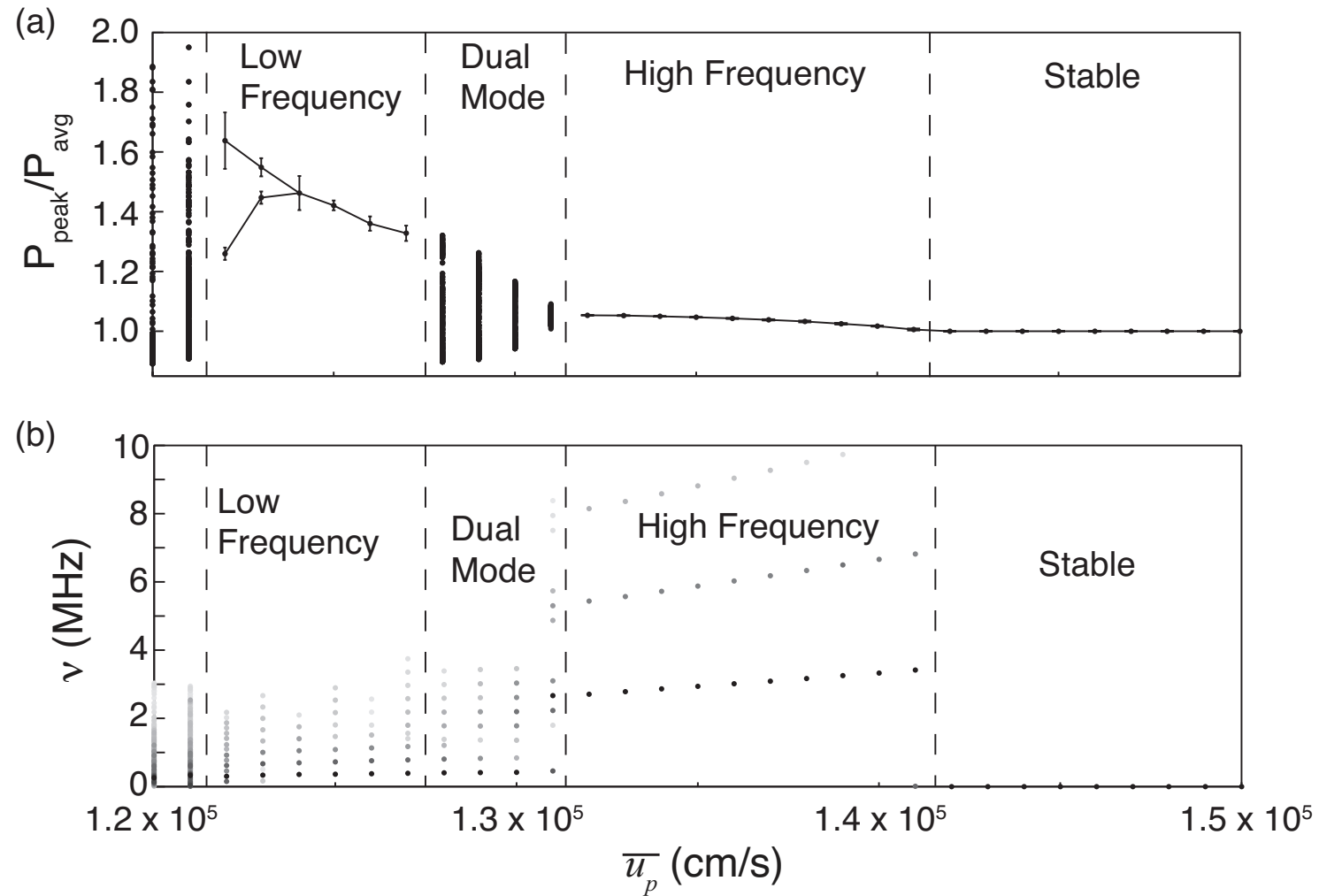


# Diffusion Reduces the Magnitude of the First and Second Harmonics





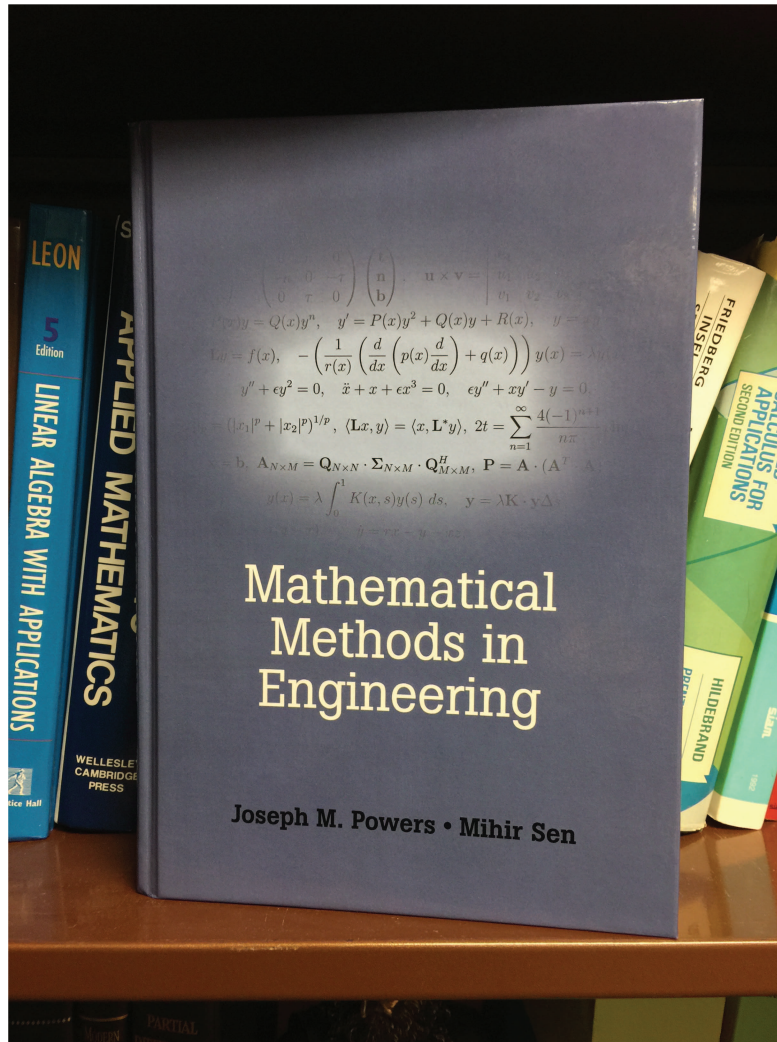
# Bifurcation Diagram for Hydrogen-Air Detonation



## Conclusions

- Long time behavior of a hydrogen-air detonation becomes more complex as the overdrive is decreased; three phenomena are predicted:
  - a stable detonation,
  - a single dominant high frequency mode oscillatory detonation,
  - a dual mode oscillatory detonation, dominated by the low frequency mode.
- Harmonic analysis has revealed the first harmonic frequency moderately lowers as the overdrive is lowered in the high frequency mode.
- At the second bifurcation there is a drastic shift in the fundamental frequency from 0.71 MHz to 0.11 MHz.
- Shock-capturing requires a four times finer grid to predict the essential dynamics of an inviscid detonation than the minimal artificial viscosity shock-fitting scheme.
- Physical diffusion causes a amplitude reduction in all cases examined; further investigation is needed at longer times near the bifurcation limits.

## A New Book



- Powers & Sen, *Mathematical Methods in Engineering*, Cambridge U. Press, 2015.
- Foundation of AME 60611, taught for over twenty-five years.
- ACMS students welcome in this course.